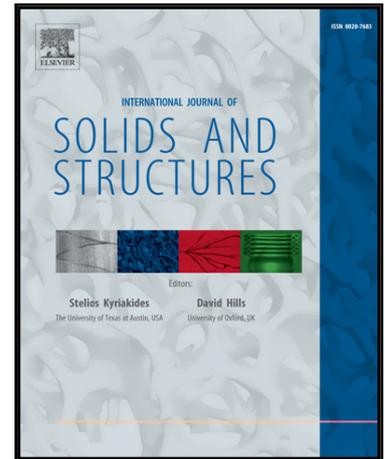


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# Damage Prediction in Single Point Incremental Forming using an Extended Gurson Model

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## Abstract

Single point incremental forming (SPIF) has several advantages over traditional forming, such as the high formability attainable by the material. Different hypotheses have been proposed to explain this behavior, but there is still no straightforward relation between the particular stress and strain state induced by SPIF and the material degradation leading to localization and fracture. A systematic review of the state of the art about formability and damage in SPIF is presented and an extended Gurson-Tvergaard-Needleman (GTN) model was applied to predict damage in SPIF through finite element (FE) simulations. The line test was used to validate the simulations by comparing force and shape predictions with experimental results. To analyze the failure prediction, several simulations of SPIF cones at different wall angles were performed. It is concluded that the GTN model underestimates the fail-

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ure angle on SPIF due to wrong coalescence modeling. A physically-based Thomason coalescence criterion was then used leading to an improvement on the results by delaying the onset of coalescence.

*Keywords:* Single Point Incremental Forming, Ductile fracture, Gurson model, Finite Element Method

*2010 MSC:* 74R99, 74S05

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## 1 **1. Introduction**

2 Nowadays, product manufacturing can be divided into two groups: rel-  
3 atively simple products manufactured in a mass production chain and spe-  
4 cialized components produced in reduced batches. Within the second group,  
5 prototyping through incremental sheet forming (ISF) has been the subject  
6 of several studies during the last decade (Jeswiet et al., 2005; Reddy et al.,  
7 2015). ISF refers to processes where the plastic deformation occurs by re-  
8 peated contact between a relatively small tool and a clamped sheet metal.  
9 The small zone submitted to plastic deformation moves during the whole  
10 process, covering the whole product and giving the final shape.

11 The focus of this work is the single point incremental forming (SPIF)  
12 process variant, where the sheet metal is deformed by a single spherical tool,  
13 which follows a complex path in order to get the required shape. One of the  
14 most prominent characteristics of the SPIF process is its flexibility. Since the  
15 shape is only given by the motion of the tool, no die is needed. The toolpath  
16 can easily be controlled using a CAD software and a change of the final  
17 shape can be quickly and inexpensively made. This dieless nature makes the  
18 SPIF process to be appropriate for prototyping, highly personalized pieces

19 and other shell-like structures. Conversely, and depending on the tool path  
20 length, the forming process can reach hours. It is, by consequence, adapted  
21 to small batch production and rapid prototyping. Applications range from  
22 large pieces like car fenders or plastic moulds, to small parts such as medical  
23 implants or prostheses.

24 Another interesting feature of the SPIF process is the deformation at-  
25 tainable by the sheet before fracture. SPIF can reach very large levels of  
26 deformation, even larger than conventional processes like the hemispherical  
27 dome (punch) test (Filice et al., 2002) or deep drawing (Jeswiet et al., 2005).  
28 The explanation of this behavior has been deeply investigated but a wide  
29 spectrum of questions still remain unanswered (Reddy et al., 2015).

30 In the present research, finite element (FE) simulations were used to pre-  
31 dict damage and fracture in the SPIF process. The article is organized as  
32 follows. Section 2 presents a literature review about damage investigations  
33 on SPIF, including some notes about the traditional formability analysis.  
34 Section 3 describes the constitutive model and the material parameters used  
35 to simulate the sheet metal. Section 4 outlines the SPIF tests used to in-  
36 vestigate damage and failure. The FE simulations and their results are also  
37 discussed here in detail. The article ends with the conclusions presented in  
38 section 6.

39 In terms of notation, the vectors and second order tensors are denoted  
40 by boldface letters, while the scalars are plain letters.  $\mathbb{H}$  represents a fourth  
41 order tensor.

## 42 2. State of the art

43 SPIF and its variants have been covered by several authors. One of  
44 the first review articles of the process was written by Jeswiet et al. (2005),  
45 covering from the experimental setup to FE analysis. Emmens and van den  
46 Boogaard (2010) published a review of technical developments on incremental  
47 forming through the years. It is important to note that the review from  
48 Emmens and van den Boogaard (2010) is more focused on ISF than SPIF.  
49 Recently, Reddy et al. (2015) reviewed SPIF concentrating their efforts in  
50 the shape accuracy and formability.

51 In this work the focus is on formability and damage leading to fracture.  
52 Formability can be understood as the ability of a material to undergo a  
53 certain amount of plastic deformation without significant damage and/or  
54 fracture. Damage prediction is linked to the formability and the deformation  
55 mechanisms, as it will be seen in the following sections.

### 56 2.1. Formability

57 As mentioned in the introduction, SPIF is characterized by an exception-  
58 ally large formality when compared to other forming processes. These obser-  
59 vations have prompted the characterization and study of the SPIF forming  
60 limits for different materials and geometries. The approach towards the un-  
61 derstanding of the increased formability can be divided in three categories:  
62 the application of formability characterization methodologies, like forming  
63 limit diagrams (FLD), the study of the effect of particular SPIF process pa-  
64 rameters on the material formability, and the prediction of rupture by FE  
65 modeling.

66 Most of the formability studies about sheet metal are rigorously embod-  
67 ied using a FLD concept, to detect a (diffuse or localized) necking condition  
68 followed by a rupture phase. FLDs were initially introduced by Keeler and  
69 Backofen (1963) and Marciniak and Kuczynski (1967). This commonly used  
70 framework has been widely adopted in the literature but suffers from im-  
71 portant drawbacks when applied to SPIF (Emmens and van den Boogaard,  
72 2009). Non radial strain paths, high stress gradients along the sheet thick-  
73 ness and the presence of through-thickness shear implying that the principal  
74 strains are not in the sheet plane are characteristics of SPIF. These specifici-  
75 ties do not respect the assumption of FLD and their use can lead to wrong  
76 conclusions (Allwood et al., 2007; Emmens and a.H. van den Boogaard, 2007).  
77 Hence, FLDs should be regarded only as an useful tool providing important  
78 insights on the material formability but not as the definitive tool to charac-  
79 terize it.

80 A short review of the mechanisms claimed to enhance SPIF formability  
81 were listed by Emmens and van den Boogaard (2009) and further detailed in  
82 Emmens (2011). An overview of some of them is given hereafter:

83 *Through-thickness shear.* In theory, under simple shear, necking is not devel-  
84 oped and rupture appears by shear band. In stretch forming, shear brings a  
85 stabilization effect by reducing the yield stress in tension, as shown by Em-  
86 mens and van den Boogaard (2009). Shear can explain the increase of the  
87 SPIF formability, as analytically shown by Allwood et al. (2007) and Eyckens  
88 et al. (2009), using an extended Marciniak and Kuczynski (1967) model.

89 *Bending-under-tension (BUT).* Also referred to as stretch bending, BUT  
90 is another stress state presenting an improved formability (Emmens and

91 van den Boogaard, 2008). It shows a considerable increase in formability  
92 when compared to cases without bending. Neglecting the stabilizing effect  
93 of bending, the conventional FLD, which assumes homogeneous stress along  
94 the thickness direction, may underestimate the forming potential. One way  
95 to overcome this drawback is to formulate the FLD in the principal stress  
96 space, instead of using (the traditional) strain-based FLD (Stoughton and  
97 Yoon, 2011).

98 *Cyclic effects.* It must be noted that during SPIF the strain history is not  
99 proportional because of successive bending and unbending around the tool.  
100 Cyclic loading, generated by serrated strain paths, has been widely observed  
101 in FE simulations within the ISF literature (Flores et al., 2007; Eyckens  
102 et al., 2007; Seong et al., 2014) but also experimentally through digital image  
103 correlation (DIC) measurements (Eyckens et al., 2010). This cyclic effect  
104 can have a great influence on formability, as demonstrated by Eyckens et al.  
105 (2007).

106  
107 In a SPIF hardware setup, several parameters were changed to identify  
108 the key parameters. The wall angle and sheet thickness seem to be the most  
109 relevant parameters in terms of formability. For a chosen material, the tool  
110 diameter and step-down size play a minor role on the forming angle (Ham  
111 and Jeswiet, 2007). As mentioned by Behera (2013), SPIF is characterized  
112 by well defined forming limits for a specific material thickness and process  
113 parameters. Hence, the maximum draw angle of SPIF cones can be used  
114 as a formability indicator. It is useful to note that failure does not take  
115 place immediately in a part with a wall angle above the failure limit; it

116 occurs at a certain depth. The stress state within a cone formed by SPIF  
117 can be linked to the *sine law* (Jeswiet et al., 2005), and it is possible for  
118 this geometry to establish a limit for thinning based on the wall angle and  
119 the initial thickness. Thus, it is straightforward to hypothesize that in order  
120 to increase the maximum wall angle, one could increase the initial thickness.  
121 However, this strategy has its practical limitations like the maximum machine  
122 load and thickness specifications of the material batch (Dufflou et al., 2008).

## 123 2.2. Damage and fracture prediction

124 Formability analysis by FLD has been for long time the traditional way to  
125 optimize the sheet metal forming operations. However, damage modeling of-  
126 fers another methodology based on the mechanisms of degradation/softening  
127 leading to final fracture. Of course, formability and damage prediction can  
128 easily be linked, however they are essentially different. Formability can be  
129 regarded as a more practical (engineering) concept. Material and process  
130 parameters generate a post-processed strain history using FLDs, while the  
131 material damage is an approach based on a particular stress or strain field  
132 histories acting in a material continuum. Damage is characterized in contin-  
133 uum mechanical models by a specific damage variable evolving until a limit  
134 is reached at the onset of crack formation (Lemaitre, 1985; Chow and Wang,  
135 1987; Voyiadjis and Kattan, 1992; Brünig, 2003). Another fundamental dif-  
136 ference between these approaches is that during damage development, the  
137 microscopic scale is not negligible, so the analysis should permanently be  
138 regarded as material dependent and needs to model the microstructure evo-  
139 lution (Garrison and Moody, 1987). The literature review shows a relatively  
140 scarce amount of SPIF research related to damage. One possible explanation

141 is that damage analysis does not often provide simplified solutions in terms of  
142 the forming process parameters. Moreover, complex damage models require  
143 complex characterization methodologies, which are not always feasible.

144 Porosity-induced damage within SPIF process has been studied, for in-  
145 stance, by Lievers et al. (2004) and Hirt et al. (2004). Lievers et al. (2004)  
146 presented a novel method to identify void nucleation parameters of a Gurson  
147 model using SPIF. This approach is sustained under the hypothesis that in  
148 some forming processes, like stretching, stretch flanging and SPIF, necking  
149 is suppressed and formability is controlled by void damage and shear band  
150 instability. Quadrangular SPIF pyramids for different aluminum alloys and  
151 wall angles were formed by Lievers et al. (2004), allowing an easy measure-  
152 ment of porosity.

153 Hirt et al. (2004) performed a simulation of a truncated pyramid formed  
154 using multi-stage forming, using a partial die. To study the stress state, the  
155 Gurson-Tvergaard-Needleman (GTN) model was used together with shell el-  
156 ements. Despite the limitations of the shell elements, the predictions showed  
157 that higher forming limits can be achieved with small forming heads and  
158 large values for the vertical pitch.

159 Silva et al. (2008) provided a theoretical model for a rotational symmetric  
160 SPIF shape, based on a membrane analysis. Sheet stretching was considered  
161 but bending and shear were neglected. It was observed that the opening mode  
162 of cracks in SPIF is similar to the one present in conventional stamping (mode  
163 I in fracture mechanics). The characterization of the stress state within the  
164 wall is given assuming plane strain condition (Filice et al., 2002; Jeswiet  
165 et al., 2005; Jeswiet and Young, 2005). In terms of damage evolution, the

166 decrease of the sheet thickness (or increase of the tool radius) shifts the Mohr  
167 circle to the tensile region, thus increasing the hydrostatic stress and the  
168 accumulated damage. This result is consistent with the findings of Hirt et al.  
169 (2004). The higher formability of SPIF process, compared to conventional  
170 stamping, is explained in terms of the meridional stress. In stamping, the  
171 level of hydrostatic stress in biaxial stretching is higher than in plane strain  
172 (and in the SPIF process), so damage grows faster.

173 Silva et al. (2011) grouped the literature review in two families: the *neck-*  
174 *ing view*, where formability is limited by necking and the raise of formability  
175 is due to stabilization mechanisms of the necking; and the *fracture view*,  
176 where formability is limited by fracture. High levels of formability come as  
177 a result of suppression of necking or low damage growth. Each view has  
178 its advantages and drawbacks. Against the necking view, it is known that  
179 forming limits in SPIF are well above conventional FLD and closer to the  
180 fracture forming limits (FFL). On the other hand, the fracture approach re-  
181 quires that all possible strains located on a specific limit to be dependent  
182 only on the material properties. Nevertheless, it is shown that the FFL can  
183 be sensitive to the tool size. Experimental studies show that the onset of the  
184 crack seem to be dependent on the formed shape. Silva et al. (2011) pro-  
185 posed a threshold where, depending on the tool radius, there is a transition  
186 between SPIF and stamping. Then crack prediction is expressed in terms  
187 of necking/suppression of necking. However, this view is not clear because  
188 localization can be a characteristic of SPIF.

189 Malhotra et al. (2012) used the Xue (2007) damage model to predict the  
190 mechanics of fracture on a SPIF cone and funnel through FE simulations.

191 The Xue (2007) model is a coupled damage model which combines plastic  
192 strain, hydrostatic pressure and shear on fracture. One of the main features  
193 of this model is that not only damage accumulation and fracture can be pre-  
194 dicted, but also the occurrence of diffused and localized necking (Xue and  
195 Belytschko, 2010). It is observed in the funnel shape that the initial damage  
196 is low due to the low initial angle and it increases dramatically until reaching  
197 an angle higher than  $70^\circ$ . It is also noticed that the shear strain is higher in  
198 the element from the inner side (i.e. the side making contact with the tool)  
199 of the sheet, delaying damage accumulation. Nevertheless, the Xue (2007)  
200 model predicts faster damage accumulation in SPIF than in deep drawing  
201 (in which the shear is small and the deformation mechanism is governed by  
202 stretching). However, failure in SPIF is greatly delayed and the sheet can  
203 achieve a larger deformation without failure than in deep drawing. Two  
204 observations can be regarded at this point. One is that from the observed  
205 thinning, plastic deformation is evenly distributed so the first localization  
206 has still to undergone neck growth when the direct tool force is already far.  
207 This fact justifies the ability of the shear band to *share* some subsequent  
208 deformation. The second observation is that if a section is still undergoing  
209 deformation after localization, it should break in this point instead of in the  
210 contact zone (as it is observed experimentally). Malhotra et al. (2012) sug-  
211 gested that since the distance from the neck to the load application increases,  
212 the ability of this neck to share deformation decreases.

213 Here, the localized effect of SPIF implies that the plastic strain is dis-  
214 tributed more evenly in the piece than in deep drawing. The already formed  
215 zone is still undergoing plastic deformation. This can explain the inability of

216 conventional FLD to predict failure in SPIF, and justifies the observed slow  
217 transition between material localization and actual fracture.

218 Summarizing, the classical way to analyze the high formability within  
219 FLD can help to understand the effect of the process parameters. For in-  
220 stance, the results obtained by FLDs suggest that through-thickness shear  
221 (TTS) is an important stabilization mechanism. However, the complexity of  
222 SPIF seems to go beyond the scope of the FLD approach. Malhotra et al.  
223 (2012) showed that TTS by itself cannot explain the high formability. Com-  
224 paring with the formability review of Section 2.1, the localized effect of BUT  
225 seems to be more important than TTS. Moreover, Silva et al. (2011) showed  
226 that the part geometry and the tool size can have a coupled effect on the  
227 formability. The effects of the thickness distribution prior to necking or fail-  
228 ure without necking are hard to capture by a classical formability analysis.  
229 Damage models, on the other hand, allow a more comprehensive understand-  
230 ing of the material behavior leading to fracture. It is not hard to observe  
231 that both approaches can be complementary. Experimental results from the  
232 FLD can be used to validate damage models.

### 233 **3. Constitutive model**

234 In this section, the constitutive models for the plastic and damage behav-  
235 ior of the material is briefly explained. The elastic part is described by the  
236 isotropic-linear version of the Hooke's law.

#### 237 *3.1. Elasto-plastic behavior*

238 The Hill (1948) yield criterion is chosen because of its overall simplicity  
239 when describing the anisotropic behavior of a metallic material. The shape

240 of this yield locus is given by the following equation:

$$\sigma_{eq} := \sqrt{\frac{1}{2} (\boldsymbol{\sigma} - \mathbf{X}) : \mathbb{H} : (\boldsymbol{\sigma} - \mathbf{X})}, \quad (1)$$

where  $\mathbb{H}$  is a fourth-order tensor containing the anisotropic parameters and  $\sigma_{eq}$  is the equivalent stress. Within the anisotropic axis frame (and omitting the backstress for the sake of simplicity), the shape of the yield locus can be written as:

$$\begin{aligned} 2\sigma_{eq}^2 := & F(\sigma_y - \sigma_z)^2 + G(\sigma_z - \sigma_x)^2 + H(\sigma_x - \sigma_y)^2 + \dots \\ & \dots + 2L\sigma_{yz}^2 + 2M\sigma_{zx}^2 + 2N\sigma_{xy}^2, \end{aligned} \quad (2)$$

241 where  $F, G, H, L, M$  and  $N$  are material parameters.

242 Assuming a *strain hardening* hypothesis, the isotropic hardening behavior  
243 can be modeled by the Swift law which shows neither saturation nor softening  
244 phenomenon:

$$\sigma_Y(\bar{\epsilon}^P) = K(\bar{\epsilon}^P + \epsilon_0)^n, \quad (3)$$

245 where  $\sigma_Y$  is the yield stress,  $\bar{\epsilon}^P$  the equivalent plastic strain and  $\epsilon_0, K, n$  are  
246 material parameters. An evolution law for the backstress tensor was proposed  
247 by Armstrong and Frederick (1966) (A-F model), including a non-linear term  
248 (Chaboche, 1977; Frederick and Armstrong, 2007):

$$\dot{\mathbf{X}} = C_X (X_{\text{sat}} \dot{\boldsymbol{\epsilon}}^P - \mathbf{X} \bar{\epsilon}^P), \quad (4)$$

249 where  $\dot{\mathbf{X}}$  is the rate of the *backstress* tensor,  $\dot{\boldsymbol{\epsilon}}^P$  is the plastic strain rate  
250 tensor.  $C_X$  (saturation rate) and  $X_{\text{sat}}$  (saturation value of the backstress)  
251 are material constants. The model is able to predict both the Bauschinger  
252 effect and accumulation of plastic strain under an asymmetrical stress cycle.

253 3.2. Damage model

254 The Gurson (1977) model is a mathematical representation of ductile  
 255 damage based on the micromechanics of the material. It is defined by an ho-  
 256 mogenization theory in the analysis of the plastic stress field in a microscopic  
 257 medium composed of a dense matrix and cavities. The model is expressed as  
 258 a macroscopic yield criterion, introducing a micromechanical variable as its  
 259 damage parameter: the *void volume fraction*  $f$ , which acts as an imperfection  
 260 during the plastic flow.

261 The Gurson-Tvergaard-Needleman (GTN) model is one of the first ex-  
 262 tensions to robustly compile the three stages of damage development: void  
 263 nucleation, growth and coalescence. The evolution of voids can be mathe-  
 264 matically assumed to be additively decomposed in a nucleation and growth  
 265 part:

$$\dot{f} = \dot{f}_n + \dot{f}_g, \quad (5)$$

266 where  $f_n$  is the nucleated void volume fraction and  $f_g$  the growth of the voids,  
 267 derived from the plastic incompressibility of the matrix:

$$\dot{f}_g = (1 - f) \operatorname{tr} \dot{\epsilon}^p. \quad (6)$$

268 Nucleation can be correlated in terms of the equivalent plastic strain in  
 269 the matrix  $\epsilon_M^P$  in the following form (Chu and Needleman, 1980):

$$\dot{f}_n = \frac{f_N}{S_N \sqrt{2\pi}} \exp \left[ -\frac{1}{2} \left( \frac{\epsilon_M^P - \epsilon_N}{S_N} \right)^2 \right], \quad (7)$$

270 where  $f_N$  is the maximum potential nucleated void volume fraction in relation  
 271 with the inclusion volume fraction,  $\epsilon_N$  is the mean effective plastic strain of

272 the matrix at incipient nucleation and  $S_N$  is the Gaussian standard deviation  
 273 of the normal distribution of inclusions.

274 The third stage, coalescence, is characterized at the macroscopic level in  
 275 a load-displacement curve by an abrupt change in the slope at the onset of  
 276 a (macroscopic) crack. In order to incorporate coalescence into the Gurson  
 277 model, Tvergaard and Needleman (1984) proposed to identify the porosity  
 278 evolution as an addition of nucleation and growth porosity rates and by a  
 279 specific coalescence function  $f^*$ , which replaces the porosity in the following  
 280 way:

$$f^* = \begin{cases} f & \text{if } f < f_{cr} \\ f_{cr} + K_f(f - f_{cr}) & \text{if } f > f_{cr} \end{cases} \quad (8)$$

281 with

$$K_f = \frac{f_u - f_{cr}}{f_F - f_{cr}}, \quad (9)$$

282 where  $f_u$  is the ultimate value of  $f^*$  at the occurrence of ductile rupture,  $f_{cr}$   
 283 is the critical void volume fraction at the onset of coalescence and  $f_F$  is the  
 284 porosity at final failure. The aim of  $f^*$  is to model the complete vanishing  
 285 of the carrying load capacity due to void coalescence.

286 The yield criterion of the GTN model introduces the factors  $q_1$  and  $q_2$  to  
 287 describe more accurately void growth mechanics (Tvergaard, 1981):

$$F_p(\boldsymbol{\sigma}, f, \sigma_Y) = \frac{\sigma_{eq}^2}{\sigma_Y^2} - 1 + 2q_1 f \cosh\left(-\frac{3q_2}{2} \frac{\sigma_m}{\sigma_Y}\right) - (q_1 f)^2 = 0, \quad (10)$$

288 where  $\sigma_m$  is the mean (hydrostatic) stress. Using a value of  $q_1 = 1.5$  and  
 289  $q_2 = 1.0$  allows the continuum model to be in good agreement with the local-  
 290 ization strain (Tvergaard, 1981).

291 *3.2.1. Thomason criterion*

292 The coalescence model in the GTN model is a purely phenomenological  
 293 approach, but physically-based coalescence criteria can also be used. For  
 294 instance, Zhang et al. (2000) incorporated a criterion based on the plastic  
 295 limit load proposed by Thomason (1990). This model has good accuracy for  
 296 both hardening and non-hardening materials. As mentioned previously, co-  
 297 alescence in the GTN model is triggered when the porosity reaches a critical  
 298 value  $f_{cr}$  and the evolution of voids is accelerated through the effective poros-  
 299 ity  $f^*$  function. The critical coalescence porosity  $f_{cr}$  is a material parameter  
 300 in the classical GTN model. In the Thomason criterion, on the contrary,  
 301 this threshold is supposed to be reached when the following inequality is no  
 302 longer satisfied:

$$\frac{\sigma_I}{\sigma_Y} < \left[ \alpha \left( \frac{1}{\chi} - 1 \right)^2 + \frac{\beta}{\sqrt{\chi}} \right] (1 - \pi\chi^2), \quad (11)$$

303 where  $\sigma_I$  is the maximum principal stress,  $\alpha$  is a material parameter defined  
 304 as a function of the hardening exponent  $n$  and  $\beta = 1.24$ . The void space ratio  
 305  $\chi$  is given by:

$$\chi = \frac{2\sqrt[3]{\frac{3f}{4\pi} \exp(\epsilon_1 + \epsilon_2 + \epsilon_3)}}{\sqrt{\exp(\epsilon_1 + \epsilon_2 + \epsilon_3 - \epsilon_{\max})}}, \quad (12)$$

306 with  $\epsilon_1$ ,  $\epsilon_2$  and  $\epsilon_3$  the principal strains, and  $\epsilon_{\max}$  the maximum principal  
 307 strain.

308 *3.2.2. Shear extension*

309 The Gurson (1977) model and the GTN extension include the triaxiality  
 310 and the mean (hydrostatic) stress as scalar parameters describing the stress  
 311 state. Nevertheless, Gologanu et al. (1996) observed that the void expansion

312 can vary in different directions under the same triaxiality. Furthermore,  
 313 the Gurson model does not behave very well under low values of triaxiality  
 314 ( $< 0.3$ ). In some cases like in shear-dominated deformations, triaxiality is  
 315 near zero or even negative predicting almost no increase of damage (in the  
 316 GTN extension of the Gurson model voids do not grow under pure shear).

317 The effect of the stress invariants on the mechanical behavior is not lim-  
 318 ited only to the use of the triaxiality or the mean stress. The third invariant  
 319 (related to the Lode angle) of the deviatoric stress has been considered in  
 320 constitutive models to predict localization (Brünig et al., 2000) and fracture  
 321 (Bai and Wierzbicki, 2008). Barsoum and Faleskog (2007) showed that the  
 322 strain localization decreases when passing from tension to shear, and the  
 323 softening rates decreases when increasing the Lode parameter. Gao et al.  
 324 (2009) demonstrated that the Lode parameter has an important effect on  
 325 the strain at coalescence and this effect is lower at high triaxiality, coinciding  
 326 with the previous results from Zhang et al. (2001).

327 Encouraged by this evidence, Nahshon and Hutchinson (2008) proposed  
 328 a shear extension for the Gurson model involving the void growth relation  
 329 (Eq. 5). Hence, the void rate is now governed by three terms:

$$\dot{f} = \dot{f}_n + \dot{f}_g + \dot{f}_s, \quad (13)$$

330 where  $f_s$  is the contribution by the shear damage. The influence of the Lode  
 331 angle is then given by:

$$\dot{f}_s = k_\omega f \omega(\boldsymbol{\sigma}) \frac{\boldsymbol{\sigma}_{\text{dev}} : \dot{\boldsymbol{\epsilon}}^P}{\sigma_{eq}}, \quad (14)$$

332 with  $\boldsymbol{\sigma}_{\text{dev}}$  the deviatoric part of the Cauchy stress tensor,  $k_\omega$  a material

333 constant and  $\omega(\boldsymbol{\sigma})$  a stress scalar function defined as:

$$\omega(\boldsymbol{\sigma}) = 1 - \left( \frac{27 J_3}{2 \sigma_{eq}^3} \right)^2; \quad 0 \leq \omega \leq 1, \quad (15)$$

334 where  $J_3$  is the third deviatoric stress invariant. This extension has however  
 335 a less straightforward link with the microstructure.  $f_s$  is more related to the  
 336 void shape and void rotation, and their impact on the stress field distribution  
 337 within the matrix. Like the coalescence extension, shear extensions are purely  
 338 phenomenological and thus the *void porosity* loses its original meaning for a  
 339 more general damage representation.

340 In Nielsen and Tvergaard (2009, 2010) it has been noted the strong con-  
 341 tribution of  $f_s$  in plane strain uni-axial tension, even if the triaxiality  $T$  is  
 342 high. A triaxiality  $T$  dependent weight function  $\Omega(T)$  is proposed. In Eq.  
 343 14,  $\omega(\boldsymbol{\sigma})$  is replaced by  $\omega_0$  defined as:

$$\omega_0 := \omega(\boldsymbol{\sigma})\Omega(T), \quad (16)$$

344 where  $\Omega(T)$  is a function which linearly decreases depending on the triaxial-  
 345 ity:

$$\Omega(T) = \begin{cases} 1 & \text{if } T < T_1 \\ (T - T_2)/(T_1 - T_2) & \text{if } T_1 \leq T \leq T_2 \\ 0 & \text{if } T > T_2 \end{cases}, \quad (17)$$

346 where  $T_1$  and  $T_2$  are material parameters.

### 347 3.2.3. Anisotropic plasticity and mixed hardening of the matrix

348 The original Gurson model is based on a development where the ma-  
 349 trix surrounding the void is perfectly plastic and obeying to the von Mises  
 350 yield criterion. Benzerga and Besson (2001) incorporated anisotropy into the

351 Gurson (1977) model and the GTN model based on experimental evidence  
 352 regarding the effect of matrix flow on particle debonding (and hence in void  
 353 evolution). This new yield criterion is defined by:

$$F_p(\boldsymbol{\sigma}, f, \sigma_Y) = \frac{\sigma_{eq}^2}{\sigma_Y^2} - 1 + 2q_1 f \cosh\left(-\frac{3q_2 \sigma_m}{\kappa \sigma_Y}\right) - (q_1 f)^2 = 0, \quad (18)$$

354 which is the same as Eq. 10 but incorporates the effect of the anisotropy  
 355 through  $\sigma_{eq}$  and the coefficient  $\kappa$ .

356 In order to introduce isotropic hardening in the matrix, an heuristic ap-  
 357 proach is followed using the Swift law (defined previously in Eq. 3). For the  
 358 kinematic hardening, classical evolution equations like the A-F model (Eq.  
 359 4) have been used previously within the Gurson model family (Mühlich and  
 360 Brocks, 2003; Ben Bettaieb et al., 2011).

### 361 3.3. Material parameters identification

362 The selected material for the experimental campaign is a DC01 steel sheet  
 363 of 1.0 mm thickness. The plastic behavior, including anisotropy and harden-  
 364 ing, is characterized by an experimental testing campaign involving homoge-  
 365 neous stress and strain fields (tensile tests in three directions, notch tensile  
 366 tests, cyclic and static shear test, microscopic investigations, etc.). Details  
 367 about the experimental tests, the identification methodology, the validation  
 368 of the identified parameters, as well as the model choice are available in  
 369 Guzmán (2016).

370 The material exhibits large ductility, being able to reach large displace-  
 371 ment before fracture and an anisotropic behavior at 45° of the RD. The  
 372 anisotropic coefficients of the Hill (1948) were identified using tensile tests

373 in three orthogonal directions plus a simple shear test. The set of plastic  
 374 parameters is given in Table 1.

Table 1: Plasticity parameters identified for the SPIF simulations.

Anisotropy coefficients	Isotropic hardening	Kinematic hardening
$F = 0.81$	$K = 542.49 \text{ MPa}$	$C_X = 113.63$
$G = 0.99$	$\epsilon_0 = 1.78 \times 10^{-2}$	$X_{\text{sat}} = 81.96 \text{ MPa}$
$H = 1.46$	$n = 0.4328$	
$N = 2.92$		

374

375 The GTN model includes several parameters of different nature. Some of  
 376 them have micromechanical roots while others are strictly phenomenological.  
 377 Hence, a methodology has been developed in order to obtain a robust set of  
 378 parameters with both numerical and physical meanings. The methodology  
 379 involves a macroscopic testing campaign with notched specimens and micro-  
 380 scopic measurements of the void volume fraction. The model parameters are  
 381 fitted to match the experimental results of force and strain field distribution  
 382 identified by DIC.

383 Taken the plasticity parameters from Table 1 (hereafter called **Swift+AF**  
 384 set) as the reference plastic parameters, Table 2 presents the model parame-  
 385 ters obtained for different extensions of the anisotropic GTN model limited  
 386 to void growth. **nuc** means that the void nucleation term  $f_n$  is added, **coa**  
 387 that the latter model is improved by the function  $f^*$  of void coalescence and  
 388 **shear** extends the **coa** model with the  $f_s$  term of shear damage, as defined  
 389 by Eq. 13-17. For further details on the plastic and damage parameter  
 390 identification, refer to Guzmán (2016).

Table 2: GTN model parameters identified for the SPIF simulations.

Set name	Nucleation				Coalescence		Shear
	$f_0$	$f_N$	$S_N$	$\epsilon_N$	$f_c$	$f_F$	$k_\omega$
nuc	0.0008	0.0025	0.175	0.42	-	-	-
coa	0.0008	0.0025	0.175	0.42	0.0055	0.135	-
shear	0.0008	0.0025	0.175	0.42	0.0055	0.135	0.25

#### 391 4. SPIF simulations

392 In all subsequent simulations, the non-linear finite element code LAGAMINE  
 393 is used. It is a lagrangian code developed by the ArGEnCo department of  
 394 the University of Liège (Cescotto and Grober, 1985). The extended GTN  
 395 model is implemented in the FE code using an implicit integration scheme  
 396 (Ben Bettaieb et al., 2011; Guzmán and Saavedra Flores, 2016).

397 Due to the important stress and strain gradients found in the sheet during  
 398 SPIF and the use of a 3D material model, the Reduced Enhanced Solid  
 399 Shell (RESS) element (Alves de Sousa et al., 2005, 2006; Ben Bettaieb et al.,  
 400 2015) is used because of its good balance between accuracy and CPU time.  
 401 This element is based on the *solid-shell* element concept, which basically lies  
 402 between a four-noded shell element and a eight-noded solid element. Hence,  
 403 it is possible to model very thin (large aspect ratio) structures using 3D  
 404 element models (like eight-noded brick elements) without any type of 2D  
 405 hypothesis (like four-noded shell elements).

406 In order to avoid locking issues, numerical techniques such as the *enhanced*  
 407 *assumed strain* (EAS) technique (Simo and Rifai, 1990), stabilization for the

408 reduced integration (Li and Cescotto, 1997) and the B-bar method (Alves de  
409 Sousa et al., 2005) are implemented at the element level.

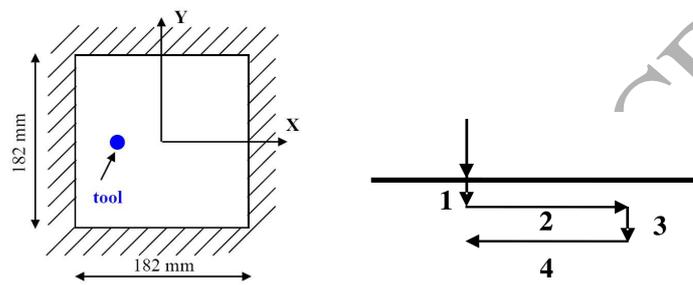
410 Contact between the tool and the sheet is modeled using the CFI3D el-  
411 ement, which is based in the penalty approach and Coulomb's friction law  
412 (Cescotto and Charlier, 1993; Habraken and Cescotto, 1998).

#### 413 *4.1. Line test*

414 The line test is one of the simplest form of SPIF. It is accurately described  
415 by Bouffioux et al. (2011). The large size of the step-down (5 mm) induces  
416 larger stress gradients than in classical SPIF problems. It allows verifying  
417 the accuracy of the identified set of material parameters and to study SPIF  
418 deformation mechanisms. The stress and strain histories during the test are  
419 similar to the ones found in SPIF test of a simple geometry. In this research,  
420 a squared sheet of 182 mm  $\times$  182 mm and 1 mm thickness of DC01 steel is  
421 clamped along its edges, as shown in Fig. 1(a). A non-rotating spindle tool  
422 of diameter 10 mm is used, following a certain toolpath (Fig. 1(b)). The  
423 test was experimentally performed at KULeuven. In order to ensure the  
424 reproducibility of the results, the whole line test was performed three times  
425 and the bolts of the frame were tightened using the same torque.

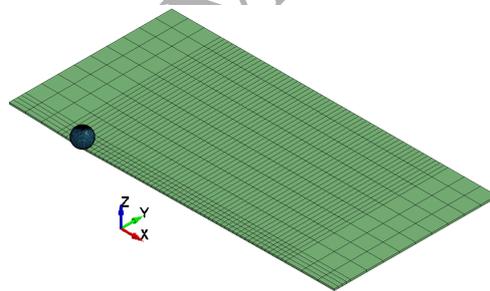
#### 426 *4.1.1. FE simulation*

427 The FE mesh is depicted in Fig. 1(c). It consists in 806 RESS solid-  
428 shell elements, one element layer with 3 integration points through-thickness  
429 and 806 CFI3D elements with 4 integration points. Symmetric boundary  
430 conditions are used along the  $X$  axis ( $Y = 0$ ) so only half of the sheet is  
431 simulated. The tool force is computed by a static implicit strategy. No



(a) Top view of the squared sheet showing the clamped edges.

(b) Tool displacement.



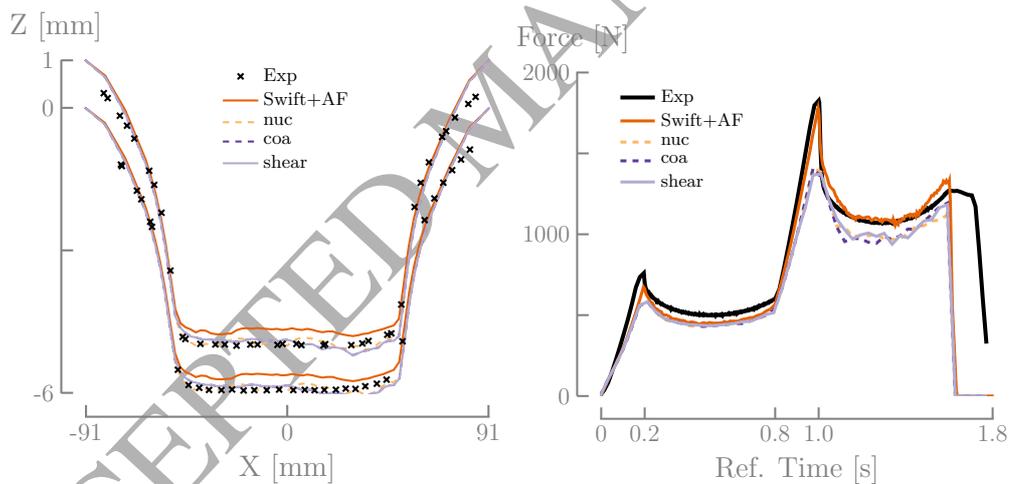
(c) FE mesh and tool tip.

Figure 1: Geometry and mesh of the line test.

432 friction is applied between the tool and the sheet.

#### 433 4.1.2. Shape and force predictions

434 An experimental-numerical comparison of four different sets of material  
 435 parameters is given in Fig. 2. The scale of the  $Z$  axis is not equal to the  $X$   
 436 axis in Fig. 2(a) in order to enhance the shape analysis, where the predicted  
 437 curves are defined by the position of the nodes located in the top and bottom  
 438 layers of the **RESS** element. The experimental results of the shape are ob-  
 439 tained through a laser line scanner mounted on the machine. Fig. 2(a) shows  
 440 the FE numerical results for a set of material parameters without damage  
 (Swift+AF) and sets considering damage (**nuc**, **coa** and **shear**). Globally,



(a) Final shape. The predicted curves are defined by the position of the nodes located in the top and bottom layers of the FE mesh.

(b) Axial force evolution.

Figure 2: Shape and force prediction for the line test and comparison with experimental results.

441

442 the predicted shapes are in good agreement with the experimental results.  
443 The predictions using the GTN model are better than those using only plas-  
444 tic parameters. Nevertheless, the differences between the predicted shape  
445 by Hill (1948) or by the GTN model and the experimental measurements  
446 are less than 0.3 mm near  $X = 0$  mm, which is small compared to the shape  
447 depth (6 mm). The difference between sets considering damage or not is due  
448 to the softening effect induced by damage. In the simulations using the GTN  
449 model, no noticeable difference is observed among **nuc**, **coa** and **shear** sets.

450 Fig. 2(b) shows the tool reaction in the  $Z$  (axial) direction during the  
451 line test. The experimental force is measured using a load cell mounted on  
452 the machine. The predictions based on the set of parameters of the damage  
453 model (**nuc**, **coa** and **shear**) are slightly lower (less than 10%) than the  
454 ones associated with the plastic model (**Swift+AF**) using mixed hardening.  
455 Again, there is no important difference among the force predictions of damage  
456 activating nucleation and coalescence steps or taking into account a shear  
457 extension.

#### 458 4.1.3. Analysis of state variables

459 The computed material state variables are analyzed within the simula-  
460 tions using the most complete GTN model (the **shear** set). The variables are  
461 retrieved from three different solid-shell elements: 118, 404 and 690, shown  
462 in Fig. 3. Element numbers 118 and 690 are located under the tool at the  
463 first (step 1 in Fig. 1(b)) and second indent (step 3), respectively. Element  
464 number 404 is located between these two elements. The results are shown in  
465 Fig. 4, where the indent step is depicted as a shaded area. The first integra-  
466 tion point (closer to the outer surface, the one not making contact with the

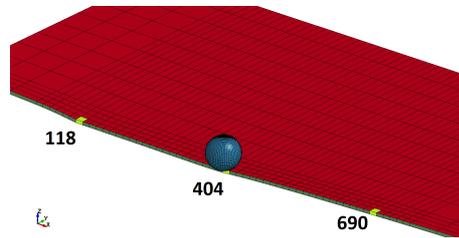
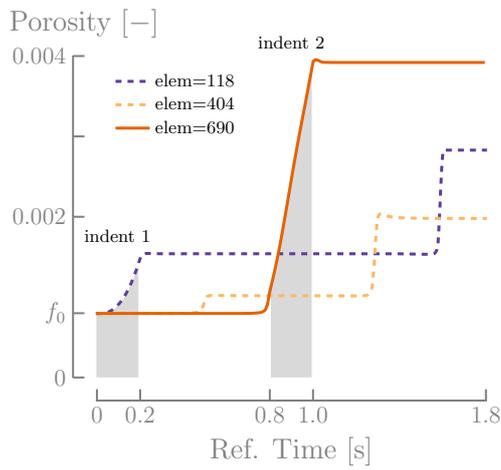


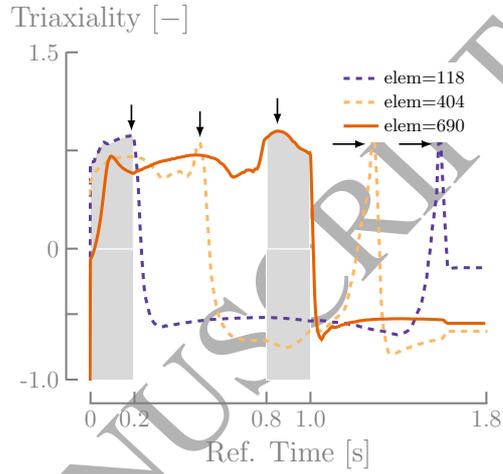
Figure 3: Line test FE mesh showing the number of elements (118, 404 and 690) selected to display state variables evolution.

467 tool) is found to give the highest equivalent plastic strain of the three inte-  
 468 gration points. This is expected, as the local stretching and bending of the  
 469 sheet around the tool causes the zone in the outer side of the sheet to stretch  
 470 more than zones in the inner side. Therefore, the state variables are analyzed  
 471 at this integration point. From Fig. 4(a) showing the effective porosity  $f^*$   
 472 evolution, it is clear that the indent steps play a major role in the porosity  
 473 history for the elements under the tool indentation (118 and 690). Element  
 474 number 404 is not affected by the tool indentation as it is too far from the  
 475 indentation zone. Nevertheless, there is a porosity increment due to the tool  
 476 contact and sheet deformation. The porosity increment after each indent  
 477 can be related with a triaxiality *peak* (marked with an arrow in Fig. 4(b)),  
 478 when the tool approaches to the element. It can be observed that triaxiality  
 479 increases when the tool approaches to the element, and decreases when the  
 480 tool moves away from the (plastically deformed) element. It must be noted  
 481 that even if the triaxiality is high for element numbers 404 and 690 during  
 482 the first indent, there is no increment of the porosity as these elements do  
 483 not deform plastically (see Fig. 4(c)) at this stage.

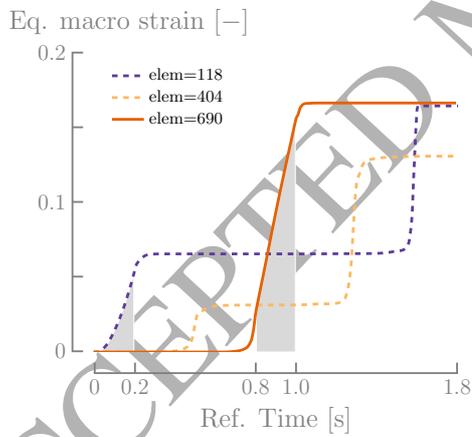
484 Triaxiality can explain why there is a porosity increment, but does not ex-



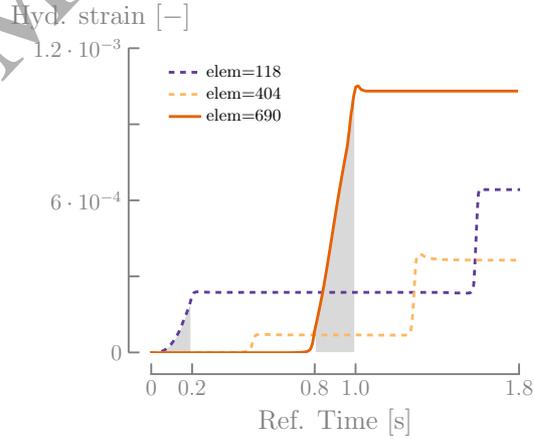
(a) Effective porosity.



(b) Triaxiality. The arrows mark triaxiality peaks.



(c) Equivalent plastic strain.



(d) Mean plastic strain.

Figure 4: State variables evolution in the line test for element numbers 118, 404 and 690. The shaded areas indicate the indent steps.

485 plain why element number 690 reaches a higher porosity than element number  
486 118, as both elements show the same level of deformation (Fig. 4(c)). The  
487 reason of this higher value is based on the mean (volumetric) plastic strain  
488 evolution shown in Fig. 4(d). It is clear that element 690 attains a higher vol-  
489 umetric strain than element 118. Therefore, the porosity mechanism during  
490 the line test is mainly governed by the triaxiality and the volumetric parts  
491 of the plastic strain. As expected, the simulation does not predict material  
492 failure as no crack appeared within the experiment. Note that the coales-  
493 cence stage is not activated within this line test, as the porosity is still far  
494 from the critical value  $f_{cr} = 0.055$  of the onset of coalescence.

#### 495 4.2. Cone test

496 Fig. 5 shows the nominal geometry of a cone of wall angle  $\alpha$  and 30 mm  
497 depth. The wall angle in this geometry is a measurement of the formabil-  
498 ity limits of SPIF for a determined material. For the DC01 steel of 1 mm  
499 thickness,  $67^\circ$  is the (experimental) maximum achievable wall angle without  
500 failure (Behera, 2013). SPIF cones with different wall angles are simulated  
501 and the porosity field is analyzed. The experimental measurements (forces  
502 and shapes) are not available for these cones, but the analytical formula of  
503 Aerens et al. (2009) is available to estimate the forming forces.

##### 504 4.2.1. FE simulation

505 Fig. 5(c) depicts a  $90^\circ$  angle pie FE mesh consisting in 1492 RESS solid-  
506 shell elements, one element layer with 3 integration points through-thickness  
507 and 1344 CFI3D elements with 4 integration points. The toolpath is com-  
508 posed of 60 contours with a step down of 0.5 mm between two successive

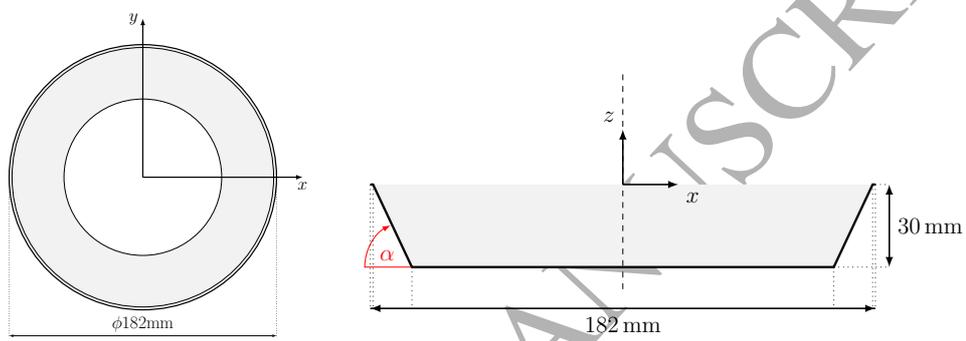


Figure 5: Geometry and mesh of the cone test.

509 contours. As the experimental cone is clamped, the nodes along the outer  
 510 circumferential part of the  $90^\circ$  pie mesh are completely fixed (in the three  
 511 translations). In the other edges, rotational boundary conditions are im-  
 512 posed. For more details about the FE model, refer to Guzmán et al. (2012b)  
 513 and Guzmán (2016).

514 Several FE simulations were carried out on SPIF cones with different wall  
 515 angles using the set **shear** from Table 2. The FE predictions of the force  
 516 are shown in Fig. 6 for four selected angles, two of them predicting material  
 failure. The GTN model predicts a failure for a  $48^\circ$  cone. The model strongly

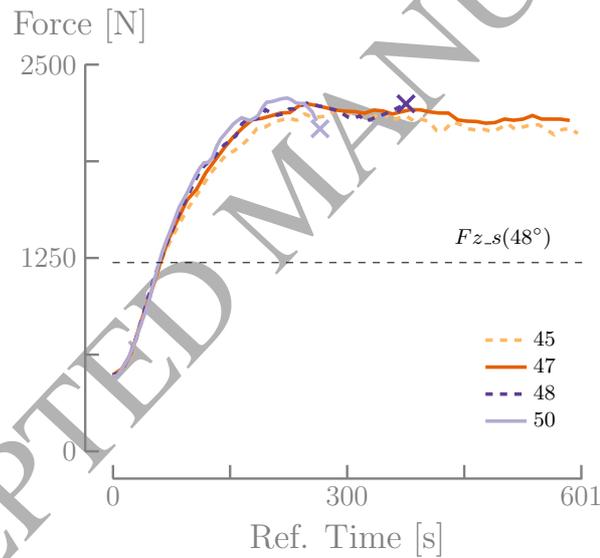


Figure 6: Axial force predictions for the cone test for different wall angles. The cross denotes the moment where  $f = f_u$  in one FE. The analytical force  $Fz_s$  predicted by the Aereens et al. (2009) formula for a  $48^\circ$  wall angle is also depicted.

517  
 518 underestimates the failure angle, since for this material and thickness, the  
 519 (experimental) critical wall angle is  $67^\circ$ . This issue is further analyzed in the

520 next section.

521 As experimental measurements are not available for this geometry, the  
522 predicted force by FE simulations is assessed using the formula proposed  
523 by Aereens et al. (2009). For a  $48^\circ$  cone the formula gives  $F_{z_s} = 1222.49$  N.  
524 Hence, the simulations overpredict the force in more than 100%. On the con-  
525 trary, the force of the line test was well predicted compared to experimental  
526 results. The only difference in terms of the FE modeling between the line  
527 and the cone test is the introduction of the rotational boundary conditions.  
528 Nevertheless, the force evolution in the cone has different characteristics than  
529 those from the line test due to different toolpath strategies. The FE formu-  
530 lation can also play a role on the force prediction. Guzmán et al. (2012a)  
531 showed using the SSH3D solid-shell element for a line test simulation, that the  
532 element flexibility modified by EAS modes can severely decrease the force  
533 level. This was later confirmed by a pyramid test simulation by Duchêne  
534 et al. (2013). Potential reasons for high forces were studied by Sena et al.  
535 (2016) (boundary conditions, missing blankholder force modeling, friction  
536 coefficient, hardening modeling choice, element stiffness, etc.). In particular,  
537 for an AA7075-O aluminum alloy using the RESS element the hardening law  
538 has an important effect on the force level. The results using the Voce law,  
539 an isotropic hardening saturation law, are better than the Swift law but still  
540 overpredicts the force. In this case, the FE force prediction for an aluminum  
541 alloys is better compared to the prediction for the steel using the same RESS  
542 FE. The accuracy of the force prediction is a classical problem in SPIF, as  
543 demonstrated by the dispersion of the simulated results of the NUMISHEET  
544 benchmark (Elford et al., 2013).

545 *4.2.2. Analysis of fracture prediction by the Gurson model*

546 It is clear in Fig. 6 that the GTN model predicts fracture at an early  
547 stage. This wrong prediction of fracture can be attributed to different factors.  
548 Two hypothesis are presented hereafter.

549 First, the predicted force level which is 100% higher than the predicted  
550 value by the Aereens et al. (2009) formula. Nevertheless, a wrong force pre-  
551 diction does not necessarily mean a wrong damage prediction. If the reaction  
552 force predicted by the FE simulations would have been the reason why dam-  
553 age increases too quickly, then the 47° cone should have failed too. Therefore,  
554 the inaccurate force prediction of the FE is not the reason of the premature  
555 failure.

556 Second, an imprecise modeling of the deformation mechanisms, such as  
557 localization and thinning, can have a critical effect on the material forma-  
558 bility. The shape and thickness distribution are correctly predicted by the  
559 RESS element, as shown in Fig. 2(a). This fact is also supported by previous  
560 simulations using the solid-shell element formulation (e.g. Duchêne et al.,  
561 2013; Sena et al., 2013). Localization is nonetheless a different aspect of  
562 the deformation. Malcher et al. (2012) showed that (in general) the GTN  
563 model does not accurately predicts the fracture strain, but it behaves rel-  
564 atively well under high and low triaxialities for the prediction of the force  
565 level and the displacement at fracture. Fig. 7 presents the equivalent plastic  
566 strain distribution for the 47° and 48° angle cones. The 47° is the limit case  
567 predicted by the model that does not fail. It is clear that strain does not  
568 localize and the plastic strain is evenly distributed, while for the 48° cone  
569 the strain localization is clear before failure. The maximum value of plastic

570 strain in Fig. 7(b) is around 0.8, which is below the usual values found on  
 571 SPIF which are easily over 1.0 (e.g. Guzmán et al., 2012b). It is possible to  
 572 observe a similar trend in the porosity distribution shown in Fig. 8. For the  
 573 48° cone, failure is preceded by localization of the equivalent plastic strain  
 574 and porosity. The 47° cone does not fail because  $f < f_F = 0.135$ , so strain  
 575 localization is triggered by the coalescence criterion of the GTN model. So,  
 576 the coalescence criterion appears as a key point that can explain the inac-  
 577 curate fracture prediction. This point is further discussed hereafter with the  
 578 effect of shear-induced damage that the classic GTN extension does not take  
 into account.

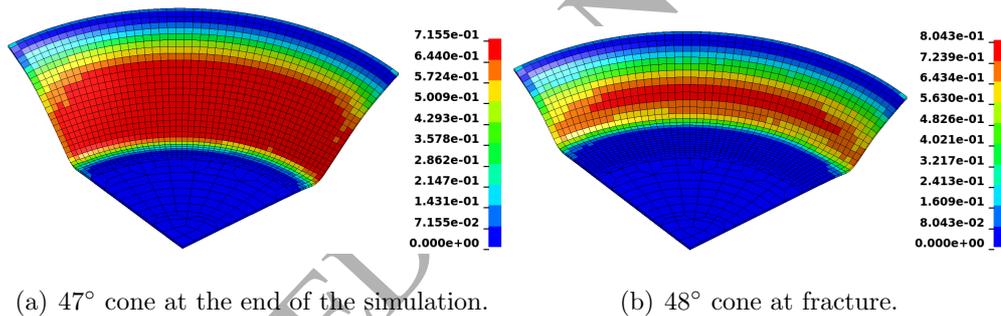


Figure 7: Equivalent plastic strain distribution for the cone test simulation.

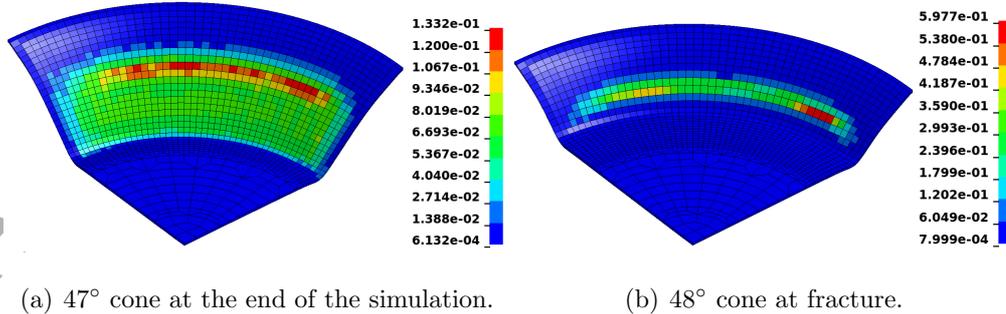


Figure 8: Effective porosity distribution for the cone test simulation.

579

580 Four variants of GTN model and coalescence are analyzed in Table 3.

581 Set `coa` is the classical coalescence model, without the shear extension. Set582 `shear` is the GTN model extended to shear. Set `coa+Thomason` and set583 `shear+Thomason` are the same as sets `coa` and `shear`, but including the

584 Thomason criterion. Table 3 presents the maximum values on the whole

585 FE mesh of the porosity reached when the coalescence starts (this value is

586 only meaningful for the Thomason coalescence criterion) and the maximum

effective porosity reached at the end of the process. It can be observed that:

Table 3: Numerical results for different types of coalescence models.

	<code>coa</code>	<code>coa+</code> Thomason	<code>shear</code>	<code>shear+</code> Thomason
Max. achievable wall angle	47°	51°	47°	51°
Max. porosity at initiation of coalescence	0.0055	0.0136	0.0055	0.0136
Max. effective porosity reached	0.1388	0.1644	0.2004	0.1546

587

588 1. The maximum achievable wall angles predicted by the variants of the

589 GTN model are significantly smaller than the experimental value.

590 2. The shear extension has a very limited influence on the results.

591 3. The Thomason coalescence criterion permitted to increase the maxi-

592 mum achievable wall angle by delaying the onset of coalescence. Indeed,

593 the porosity attained when the Thomason criterion is no longer fulfilled

594 is way larger than the parameter  $f_{cr}$  of the classical GTN model.

595 4. The maximum effective porosity exceeded the failure limit  $f_F$ . How-  
 596 ever, such values only appear very locally in the simulations. These  
 597 values were not considered to be associated with failure in this research.

598 Fig. 9 presents numerical results when the failure is predicted (i.e. when  
 599 the maximum achievable wall angle is exceeded by  $1^\circ$ ) for the GTN+Shear+Thomason  
 600 variant. It appears that that the porosity reaches large values only in a zone  
 601 around the final path of the tool. Locally, the porosity can be significantly  
 602 larger than the failure limit. According to Fig. 9(b), coalescence appears in  
 a similar zone.

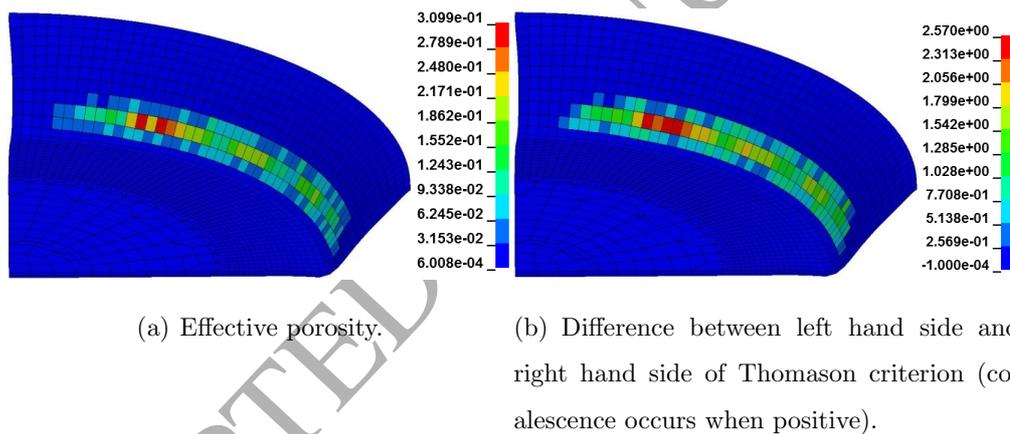


Figure 9: Numerical results for the GTN+Shear+Thomason model when failure is reached (the wall angle is  $52^\circ$ ).

603

## 604 5. Gurson versus continuum approach

605 Summarizing, the most probable reason of the premature prediction of  
 606 material failure by the GTN model is an inadequate coalescence criterion.  
 607 Indeed, it has often been discussed that  $f_{cr}$  is not a sufficient criterion to

608 describe the initiation of fracture (e.g. Malcher et al., 2014). Triggering  
609 failure based only in the damage parameter (effective porosity) could be risky  
610 considering the complexity of the stress and strain path found on SPIF.

611 To further analyze the fracture prediction of the GTN model, a com-  
612 parison will be given with other damage models. Ben Hmida (2014) used  
613 a Lemaitre type damage model in LS-Dyna explicit using a solid element.  
614 The identification of the elasto-plastic and damage parameters follows a two  
615 step procedure. Inverse analysis was used on a tensile test for elastoplastic  
616 parameters and then in a micro-SPIF test for the damage parameters. The  
617 simulations are able to predict the force level and failure in a micro-SPIF  
618 pyramid frustums. The Lemaitre model is based on the strain equivalence  
619 principle, which establishes a coupling between hardening and the damage  
620 variable. Malhotra et al. (2012) used a fracture model developed by Xue  
621 (2007). This model leads to good results when predicting the force and the  
622 depth at which fracture happens. In the Xue (2007) model, the damage evo-  
623 lution is function of the ratio of plastic strain and the fracture strain (the  
624 *self-similarity* hypothesis). In both cases, the main difference with the GTN  
625 model is that the damage models present a coupling between damage and the  
626 plastic strain. Originally, the Gurson (1977) model was developed to repre-  
627 sent the deterioration of a porous material, based on unit-cells calculations.  
628 On the contrary, the Xue (2007) model is based on a theory where the plastic  
629 damage incorporates all the three stress invariants.

## 630 6. Conclusions and perspectives

631 In this paper, an evaluation of the GTN model extended to shear is per-  
632 formed. The effects of the Thomason coalescence criterion are also checked.  
633 A review of the state-of-the-art about formability and damage in SPIF is  
634 also presented. The line test is used to validate the simulations by compar-  
635 ing force and shape prediction with experimental results. In general, the  
636 results of the shape prediction are in good agreement with the experimen-  
637 tal results. The fracture detection is correct for plane tests, while for SPIF  
638 the rupture associated to an angle of  $67^\circ$  is strongly underestimated. The  
639 good results obtained for the line tests are, unfortunately, not repeated on  
640 more complex shapes like the cone. For example, the force prediction is too  
641 high compared to experimental values, probably because of the boundary  
642 conditions. This is an issue that requires more research, as the deformation  
643 mechanisms are highly dependent on the process parameters so conclusions  
644 derived from some geometries are not necessarily repeatable in other shapes.  
645 On the other hand, the GTN model is capable to detect failure in a cone  
646 test, but the prediction is too premature compared to the experimental fail-  
647 ure angle for the same material and geometry. After performing several FE  
648 simulations of SPIF cones with different wall angles, it is concluded that the  
649 GTN model underestimates the reference failure angle. The most probable  
650 reason for an imprecise failure modeling is the coalescence model, which de-  
651 pends only on the damage parameter (porosity). Moreover, the GTN model  
652 uncouples this damage parameter with hardening. Other models like the one  
653 proposed by Xue (2007) or the Lemaitre model used by Ben Hmida (2014),  
654 which predicts failure in the SPIF process, couple the damage evolution and

655 failure with the plastic strain. This research indicates that the developed  
656 failure mode cannot be predicted by the classical assumptions of the GTN  
657 model. Even if the damage model is capable to predict the loss of the loading  
658 capacity for notched specimens, the stress and strain path found on SPIF are  
659 different and certainly more complex.

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