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Mechanical Characterization of Nodular Ductile Iron

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Summary

The objective of this study is to characterize the strength and fracture response of nodular ductile iron (NDI) and its underlying ferritic matrix phase. Quasistatic and split Hopkinson pressure bar (SHPB) compression tests were performed on NDI and a model material for the NDI matrix phase (Fe-Si alloy). Smooth and notch round bar (NRB) samples were loaded in tension until fracture to determine strain-at-failure with varying stress triaxiality. Multiple tests were performed on each small and large smooth bar samples to obtain fracture statistics with sample size. Fracture statistics are important for initializing simulations of fragmentation events. Johnson-Cook strength models were developed for the NDI and the Fe-Si alloy. NDI strength model parameters are: $A=525$ MPa, $B=650$ MPa, $n=0.6$, and $C=0.0205$. The average SHPB experimental strain-rate of 2312/s was used for the reference strain-rate in this model. Fe-Si alloy strength model parameters are: $A=560$ MPa, $B=625$ MPa, $n=0.5$, and $C=0.02$. The average SHPB experimental strain-rate of 2850/s was used for the reference strain-rate in this model. A Johnson-Cook failure model was developed for NDI with model parameters: $D_1=0.029$, $D_2=0.44$, $D_3=-1.5$, and $D_4=D_5=0$. An exponential relationship was developed for the elongation-at-failure statistics as a function of length-scale with model parameters: $S_{f1}=0.108$, $S_{f2}=-0.00169$, and $L_m=32.4$ μm . NDI strength and failure models, including failure statistics, will be used in continuum-scale simulations of explosively-driven ring fragmentation. The Fe-Si alloy strength model will be used in mesoscale simulations of spall fracture in NDI, where the NDI matrix phase is captured explicitly.

Methods

Materials

Nodular ductile iron (NDI) is a Fe-C-Si alloy and its chemical composition by weight is 93.3% Fe, 3.7% C, 2.5% Si, 0.3% Mn. NDI also has trace amounts of P, S, Cr, and Mg. NDI comprises a ductile matrix (first phase) with loosely-bound, graphite particles (second-phase) that are central to its failure response [1-3]. NDI is also an ideal material for studying the influence of microstructure on ductile fracture because it contains a readily identifiable second-phase particle population, embedded in a ductile metallic matrix, which serves as primary void nucleation sites. The material in this study met ASTM A536, 60-40-18 specifications (minimum 60 ksi (413 MPa) ultimate strength, minimum 40 ksi (275 MPa) yield strength, 18% elongation in 2 inch bar) and was purchased from DuraBar®. A Fe-2.5%Si alloy was used to model the matrix phase in NDI. The Fe-Si alloy was purchased from Carpenter Technology.

Split Hopkinson Pressure Bar Compression Experiments

As part of this study, split Hopkinson pressure bar (SHPB) compression experiments [4-5] were conducted to determine the strain-hardening response of NDI and its underlying Fe-Si matrix phase at strain rates of 2,000-4,000 s^{-1} . The test uses a cylindrical projectile to impact a cylindrical incident bar which has been instrumented with strain gages. **Figure 1** shows a schematic of the SHPB experiment. The impact sends a compressive stress wave down the incident bar, into the test sample and out through a transmitter bar with similar instrumentation. The signals from the two sets of instrumentation are compared and the net change is used to calculate the stress and strain caused by the stress wave.

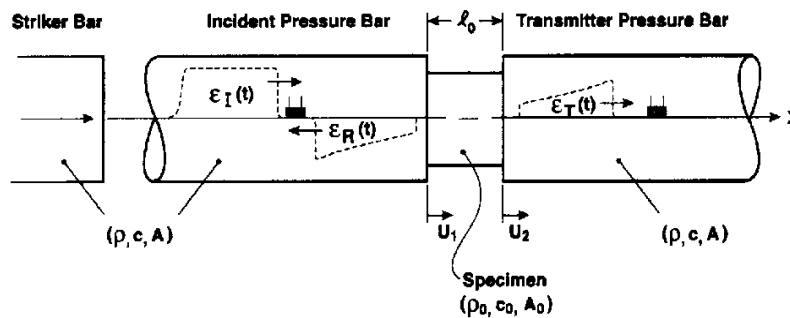


Figure 1. Schematic of split Hopkinson pressure bar (SHPB) compression experiment with emphasis on sample loading.

Figure 2 shows the SHPB compression experiments performed at LLNL. It employs a modified Dynapak pneumatic press to supply the impulse to accelerate the cylindrical projectile with a maximum velocity of approximately 900 in/s (22.9 m/s). The standard compression bar set (incident and transmitter bars) used is made from high strength steel to provide for a wide range of material testing conditions. We normally use 0.2 in (0.508 cm) height by 0.2 in (0.508 cm) diameter cylindrical test samples which fit the 0.56 in (1.42 cm) steel compression bar set. Smaller sample sizes can be accommodated in cylindrical configurations as needed. The strain signals from the two bars during test are captured with a Nicolet Integra system at rates from 1 KHz to 20 MHz and evaluated using the commercial spreadsheet software to produce graphical representations of stress and strain.



Figure 2. LLNL SHPB compression experiments.

SHPB experiments are essentially one dimensional uniaxial (stress) compression experiments. The one dimensional wave equation is used to develop equations relating the strain gage data to the specimen stress and strain rate in SHPB compression experiments. Starting with equation (1), the displacement gradient and rate can be defined at each incident bar (subscript 1) and transmitter bar (subscript 2):

$$u_1 = f(x - c_b t) + g(x + c_b t) \quad u_2 = h(x - c_b t) \quad (1.)$$

$$\dot{u}_1 = c_b (-f' + g') = c_b (-\varepsilon_i + \varepsilon_r) \quad \dot{u}_2 = -c_b h' = -c_b \varepsilon_t \quad (2.)$$

$$\frac{\partial u_1}{\partial x} = f' + g' = \varepsilon_i + \varepsilon_r \quad \frac{\partial u_2}{\partial x} = h' \quad (3.)$$

where, ε_i is the incident strain, ε_r is the reflected strain, and ε_t is the transmitted strain. The strain data is taken directly from strain gages affixed to the incident and transmitter bar. The specimen strain rate and stress can now be defined:

$$\dot{\varepsilon}_s = \frac{1}{l_0}(\dot{u}_1 - \dot{u}_2) = \frac{c_b}{l_0}(-\varepsilon_i + \varepsilon_r + \varepsilon_t) \quad (4.)$$

$$\sigma_s = \frac{EA_0(\varepsilon_i + \varepsilon_r + \varepsilon_t)}{2A} \quad (5.)$$

where, ε_s is the specimen strain rate, l_0 is the instantaneous specimen length, σ_s is the specimen stress, E is the Young's modulus of the incident and transmitter bars, A_0 is the initial cross-sectional area of the incident and transmitter bars, and A is the instantaneous specimen cross-sectional area. Since there is a finite distance between incident and transmitter bar strain gages, ε_i , ε_r and ε_t are out-of-phase. Strain gage data must be aligned before strain rate or stress is calculated, as part of the "three-wave" analysis to obtain the material strength response.

Quasistatic Tension Fracture Experiments: Statistics at Different Length-Scales

Quasistatic tension fracture experiments with two configurations of smooth round specimens were performed to probe variations of elongation-at-failure for differing size. **Figure 3** shows the 0.5 in (1.27 cm) and 0.113 in (0.287 cm) diameter specimen configurations. Approximately twenty replicate tests were performed for each specimen configuration in order to obtain adequate statistics. Experiments were performed using the UC Davis 20 kip Instron 1331 system. Elongation was measured using a MTS 634.25 axial extensometer, shown in **Figure 4**. Experiments conformed to ASTM E8 standards.



Figure 3. Smooth round NDI specimens- 0.5 in (1.27 cm) and 0.113 in (0.287 cm) diameter configurations.

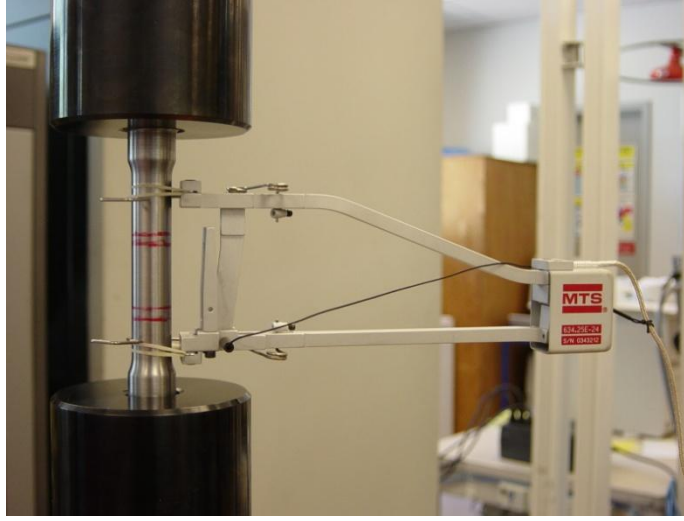


Figure 4. Measuring elongation in smooth round specimens with MTS 634.25 axial extensometer.

In order to relate one dimensional engineering stress and strain measurements to true stress and strain, as well as other convenient parameters, we use the following equations,

$$\begin{aligned}
 s &= \frac{P}{A_0} \\
 e &= \frac{L - L_0}{L_0} \\
 \sigma &= \frac{P}{A} \\
 \sigma &= s(e + 1) \\
 \varepsilon &= \ln(e + 1) = \ln\left(\frac{L}{L_0}\right)
 \end{aligned}
 \tag{6.}$$

where s is the engineering stress, e is the engineering strain, σ is the true stress, ε is the true strain, P is the load, L_0 and L are the initial and current lengths, and A_0 and A are the initial and current cross-sectional areas. *Relationships in Equation (6) are applicable only until the onset of necking.* After necking, deformation is no longer homogeneous, but localized, along the specimen and the isochoric assumption no longer applies.

Quasistatic Notched Round Bar (Tension) Fracture Experiments

Quasistatic tension fracture experiments were performed with three configurations of notched round bar (NRB) specimens to explore the influence of stress triaxiality (σ_m / σ_{eff}) on strain-at-failure in NDI. Stress triaxiality was varied by changing the notch length, 2ρ , of the specimen to 0.0564 in (0.143 cm), 0.113 in (0.287 cm), and 0.226 in (0.574 cm), while keeping the diameter, $2a$, fixed at 0.113 in (0.287 cm). **Figure 5** shows these three notched round bar configurations and a schematic with the key notch geometric parameters, $2a$ and 2ρ . Also, fifteen replicate tests were performed for each specimen configuration in order to probe strain-at-failure variations at different stress triaxialities. Experiments were performed using the LLNL 5 kN MTS system. Strain-at-failure was measured using a custom (flexure) diametral extensometer located at the notch center.

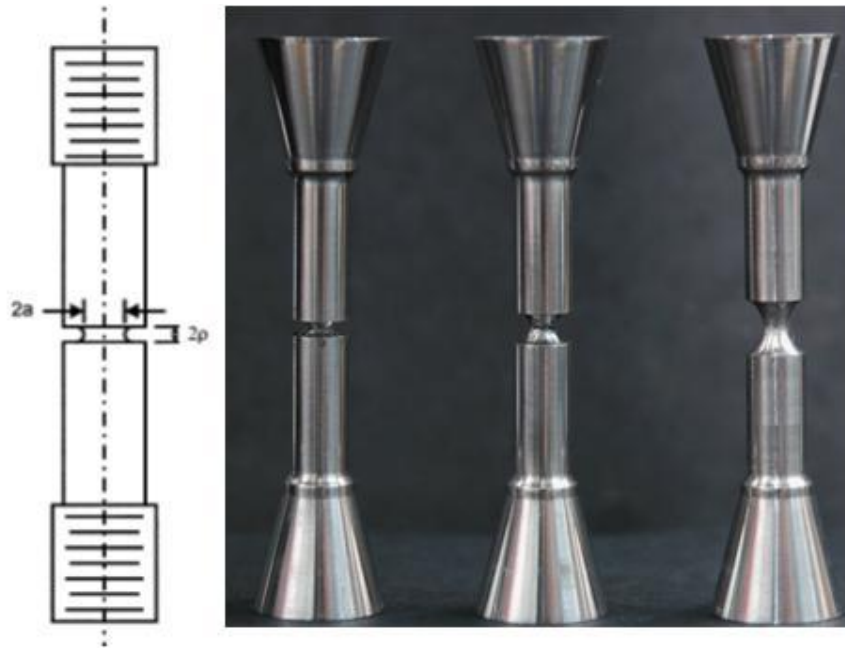


Figure 5. Notched round bar schematic and actual NDI specimens with fixed notch diameter of 0.113 in (0.287 cm) and notch lengths of 0.0564 in (0.143 cm) (left), 0.113 in (0.287 cm) (middle), and 0.226 in (0.574 cm) (right).

The (initial) stress triaxiality of the NRB specimens can be calculated using the Bridgman analysis [6],

$$\frac{\sigma_m}{\sigma_{eff}} = \frac{1}{3} - \ln\left(\frac{a}{2\rho} + 1\right) \quad (7.)$$

where σ_m is the mean stress (positive in tension), σ_{eff} is the effective stress (e.g., von Mises stress), a is the notch radius, and 2ρ is the notch length. Based on Equation (7), the initial stress triaxialities for the three specimen configurations are: 0.556 ($2\rho = 0.226$ in), 0.739 ($2\rho = 0.113$ in), and 1.02 ($2\rho = 0.0564$ in). The equivalent plastic strain (EPS), $\bar{\epsilon}^{pl}$, was calculated based on direct diametric measurements using the relationship,

$$\bar{\epsilon}^{pl} = 2 \ln\left(\frac{D_o}{D}\right) \quad (8.)$$

where D_o and D are the initial and current diameters. Equation (8) was used to calculate the EPS at failure for the different NRB specimen configurations.

Results

NDI Strength at Different Strain-Rates

Figure 6 shows the true stress versus EPS for NDI at quasistatic (0.00068/sec average) and higher-rates (2312/sec average) with a fit to the Johnson-Cook strength model [7]. This is a five-parameter empirical model that describes the yield strength as a function of plastic strain, plastic strain rate, and homologous temperature. It is expressed as,

$$\sigma = \left(A + B \left(\bar{\epsilon}^{pl} \right)^n \right) \left(1 + C \ln \left(\frac{\dot{\bar{\epsilon}}^{pl}}{\dot{\epsilon}_0} \right) \right) \left(1 - \left(\frac{T - T_{room}}{T_{melt} - T_{room}} \right)^m \right) \quad (9.)$$

where A , B are the strain-hardening coefficients, n is the strain hardening exponent, $\bar{\epsilon}^{pl}$ is the current EPS, C is the strain-rate hardening coefficient, $\dot{\epsilon}_0$ is the reference strain rate (set to 2312/sec, which is average rate from SHPB experiments), $\dot{\bar{\epsilon}}^{pl}$ is the current EPS rate, T_{room} is the room (reference) temperature, T_{melt} is the melt temperature, m is the temperature exponent, and T is the current temperature. **Table 1** summarizes the Johnson-Cook strength model parameter values, as well as the Young's Modulus which was calculated from a linear fit to quasi-static strength data up to a strain of 0.001. Since all experiments were performed at room temperature, i.e., no temperature dependence was determined, the temperature exponent was set to 1 and the melt temperature set to large value (100,000 K).

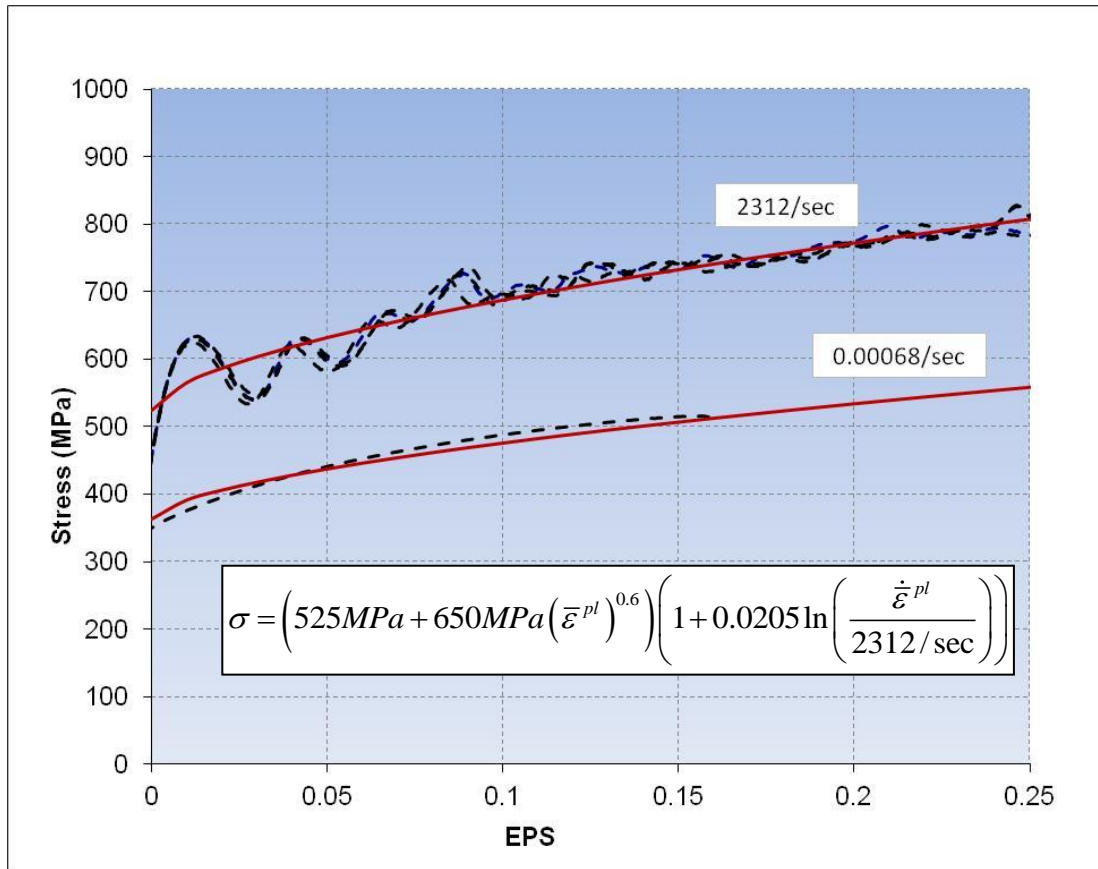


Figure 6. True stress versus EPS for NDI at quasistatic and higher-rates (black dashed lines). Johnson-Cook strength model fit also shown (red solid lines).

Table 1. Young's Modulus and Johnson-Cook strength model parameters for NDI

| E (GPa) | A (MPa) | B (MPa) | n | C | $\dot{\epsilon}_0$ (1/s) | T_{room} (K) | T_{melt} (K) | m |
|------------|------------|------------|-----|--------|-----------------------------|-------------------|-------------------|-----|
| 151.1 | 525 | 650 | 0.6 | 0.0205 | 2312 | 298 | 100,000 | 1.0 |

Fe-Si Alloy Strength at Different Strain-Rates

Figure 7 shows the true stress versus strain of the Fe-Si alloy at quasistatic (0.00089/sec average) and higher-rates (2850/sec average) with Johnson-Cook strength model fit. **Table 2** summarizes the Johnson-Cook strength model parameter values, as well as the Young's Modulus which was provided by the material vendor (Carpenter Technology). As with NDI, the temperature exponent was set to 1 and the melt temperature set to large value (100,000 K).

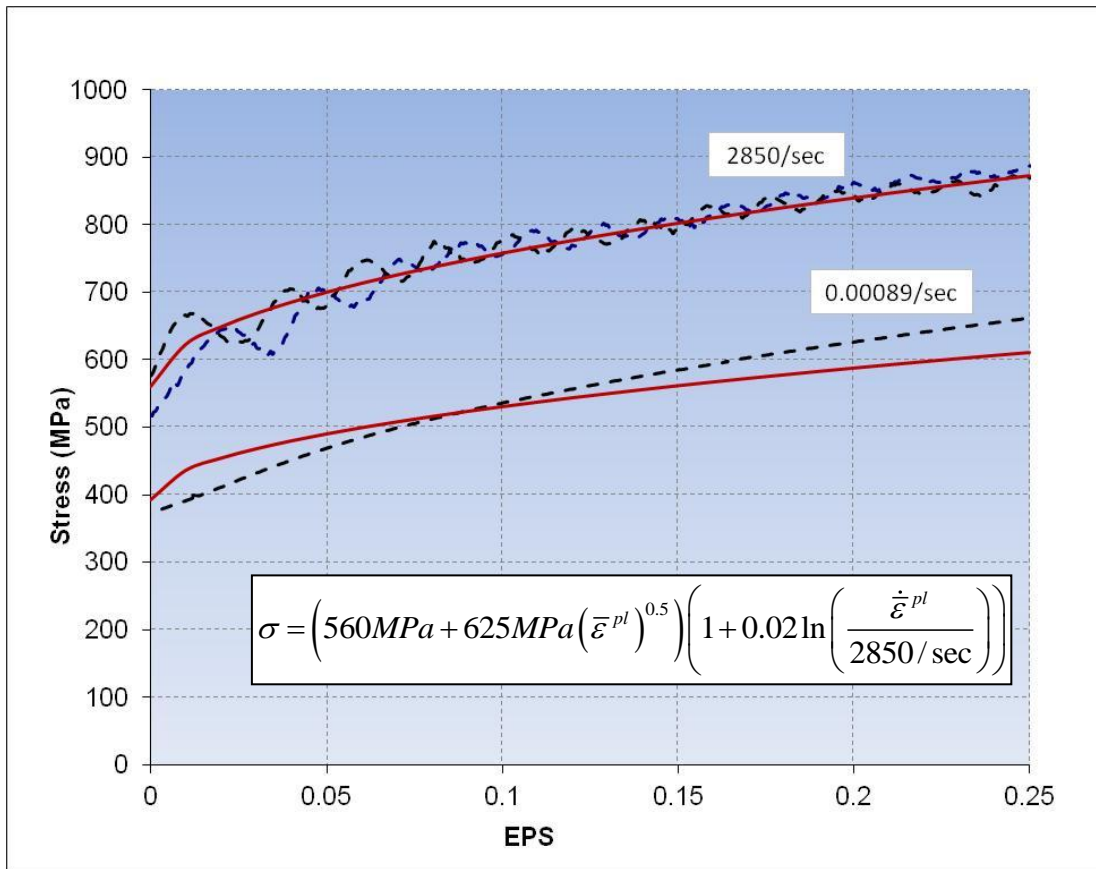


Figure 7. True stress versus EPS for Fe-Si alloy at quasistatic and higher-rates (black dashed lines). Johnson-Cook strength model fit also shown (red solid lines).

Table 2. Johnson-Cook strength model parameters for the Fe-Si alloy

| E (GPa) | A (MPa) | B (MPa) | n | C | $\dot{\epsilon}_0$ (1/s) | T_{room} (K) | T_{melt} (K) | m |
|------------|------------|------------|-----|------|-----------------------------|-------------------|-------------------|-----|
| 167 | 560 | 625 | 0.5 | 0.02 | 2850 | 298 | 100,000 | 1.0 |

NDI EPS-at-Failure for Different Stress Triaxialities

Figure 8 shows the EPS at failure for the NRB experiments as a function of stress triaxiality with a fit to the Johnson-Cook fracture strain model [8]. This is a five-parameter empirical model that describes the fracture strain as a function of stress triaxiality, strain rate, and homologous temperature. It is expressed as,

$$\bar{\epsilon}^f = \left(D_1 + D_2 \exp \left(D_3 \frac{\sigma_m}{\sigma_{eff}} \right) \right) \left(1 + D_4 \ln \left(\frac{\dot{\epsilon}^{pl}}{\dot{\epsilon}_0} \right) \right) \left(1 + D_5 \left(\frac{T - T_{room}}{T_{melt} - T_{room}} \right) \right) \quad (10.)$$

where D_1 to D_5 are fitting constants and the other terms have been defined previously for the Johnson-Cook strength model. The first part of Equation (10) that describes the stress triaxiality is a generalization of the Hancock-McKenzie model [9]. As with [9], we assume that D_3 is equal to -1.5 based on theoretical models of spherical void growth. Since we conducted all the experiments at a fixed strain rate (quasistatic) and fixed (ambient) temperature, we zero the strain-rate and temperature terms. It should be noted that the $\sigma_m = -p$ since pressure in the hydrocodes is defined to be positive in compression. So, the sign for D_3 may need to be changed depending on how this failure model is implemented in the hydrocode. **Table 3** summarizes the Johnson-Cook failure strain model parameter values. Data for the smooth bar specimens are not included in this section since only elongation-at-failure, not EPS-at-failure, was measured for those experiments.

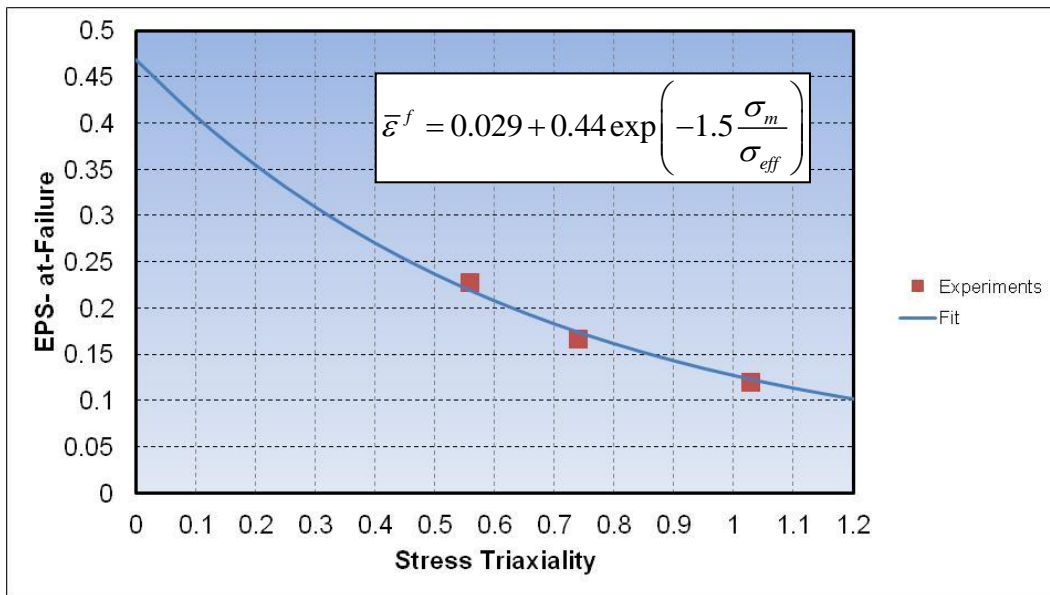


Figure 8. EPS at failure versus stress triaxiality data for NDI for NRB specimens (red boxes). Johnson-Cook failure strain model fit also shown (blue line).

Table 3. Johnson-Cook failure strain model parameters for NDI. All parameters are unitless.

| D ₁ | D ₂ | D ₃ | D ₄ | D ₅ |
|----------------|----------------|----------------|----------------|----------------|
| 0.029 | 0.44 | -1.5 | 0 | 0 |

Table 4 shows the mean and standard deviation of EPS at failure for the different specimen geometries (stress triaxialities). The basis for failure statistics are the (fifteen) replicate tests performed in each configuration. The coefficient-of-variation (COV) is the ratio of the standard deviation (SD) to the mean value and is used here to normalize the SD values.

Table 4. EPS at failure statistics for different stress triaxialities (15 samples).

| σ_m / σ_{eff} | Mean and Standard Deviation, EPS at Failure | Coefficient-of-Variation |
|---------------------------|---|--------------------------|
| 0.556 | 0.229±0.00933 | 0.0407 |
| 0.739 | 0.168±0.00528 | 0.0314 |
| 1.02 | 0.121±0.00445 | 0.0367 |

NDI Elongation Statistics at Different Length-Scales

Table 5 shows the elongation-at-failure (e_f) statistics for 1.27 cm and 0.287 cm diameter NDI (smooth, round bar) specimens. The mean and standard deviation, as well as the COV, are calculated for e_f .

Table 5. NDI elongation-at-failure statistics for 1.27 cm and 0.287 cm specimens.

| Diameter (cm) | Number of Specimens | e_f Mean and SD | e_f COV |
|---------------|---------------------|-------------------|-----------|
| 0.287 | 21 | 0.1856±0.0173 | 0.0930 |
| 1.27 | 19 | 0.1896±0.0106 | 0.0557 |

Based on this data, we develop an exponential relationship for the COV_f as a function of length-scale similar to that in [10]:

$$COV_f(L) = \left[S_{f1} \exp\left(S_{f2} \frac{L}{L_m}\right) \right] \quad (11.)$$

where L is the specimen length-scale (using specimen diameter) and L_m is a microstructural length-scale. Based on [10], we use the graphite nodule nearest neighbor distance (32.4 μm) as the microstructural length-scale. Fitted parameters S_{f1} and S_{f2} are 0.108 and -0.00169, respectively. Relationships like Equation (11) enable the simulation of natural fragmentation events, which require material failure statistics.

Conclusions

The strength and fracture properties of nodular ductile iron (NDI), as well as its underlying ferritic matrix phase (Fe-Si alloy), were characterized in this study. The strength of NDI and the Fe-Si alloy were found to be dependent on equivalent plastic strain and plastic strain rate. Reasonable fits to NDI and Fe-Si alloy data were demonstrated using the Johnson-Cook strength model. The equivalent plastic strain at failure in NDI was found to be dependent on stress triaxiality. A reasonable fit to experimental data was demonstrated using the Johnson-Cook failure strain model. The length-scale dependent elongation-at-failure statistics were fit to an exponential model. NDI strength and failure models, including failure statistics, will be used in continuum-scale simulations of explosively-driven ring fragmentation. The Fe-Si alloy strength model will be used in mesoscale simulations of spall fracture in NDI, such as [10], where the NDI matrix phase is captured explicitly.

Acknowledgements

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