

# Improving Durability of Turbine Components Through Trenched Film Cooling and Contoured Endwalls

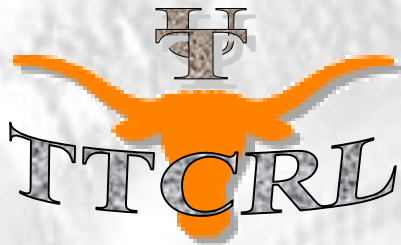
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**UTSR Peer Review Workshop  
October 20, 2010**

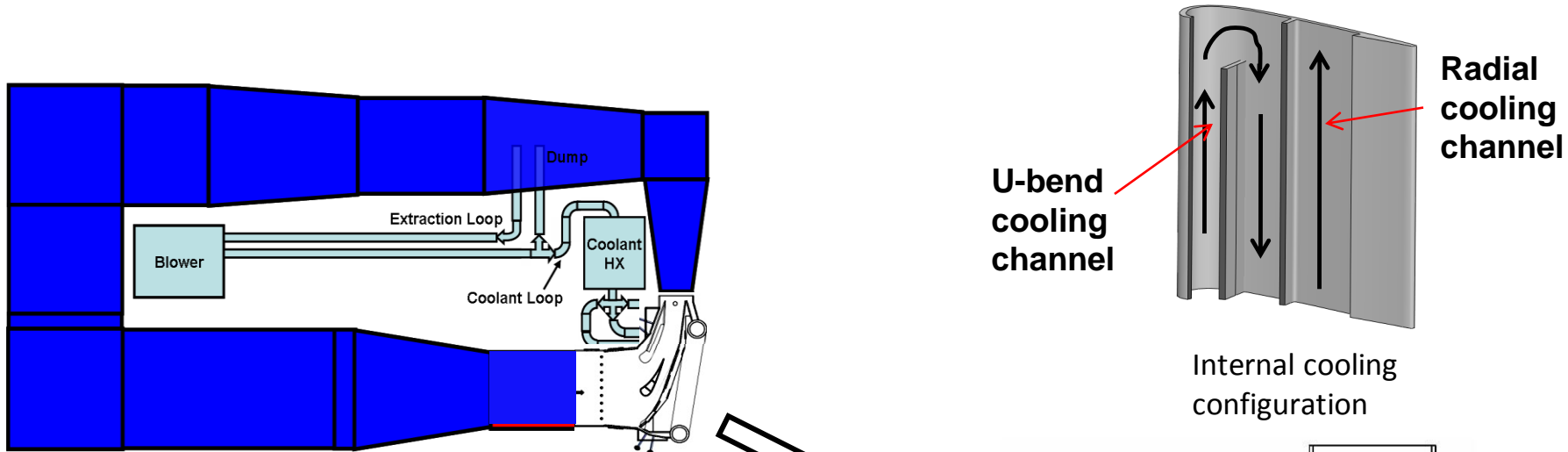
# Project Objectives

- a. Evaluate the degradation of performance for trench and crater film cooling configurations when subjected to active deposition of contaminants. This will be done on simulated vane and endwall models.
- b. Design improved trench or crater film cooling configurations that mitigate the degradation effects of deposition of contaminants.
- c. Determine the overall cooling effectiveness (including conjugate heat transfer effects) with and without thermal barrier coatings for the vane and endwall.

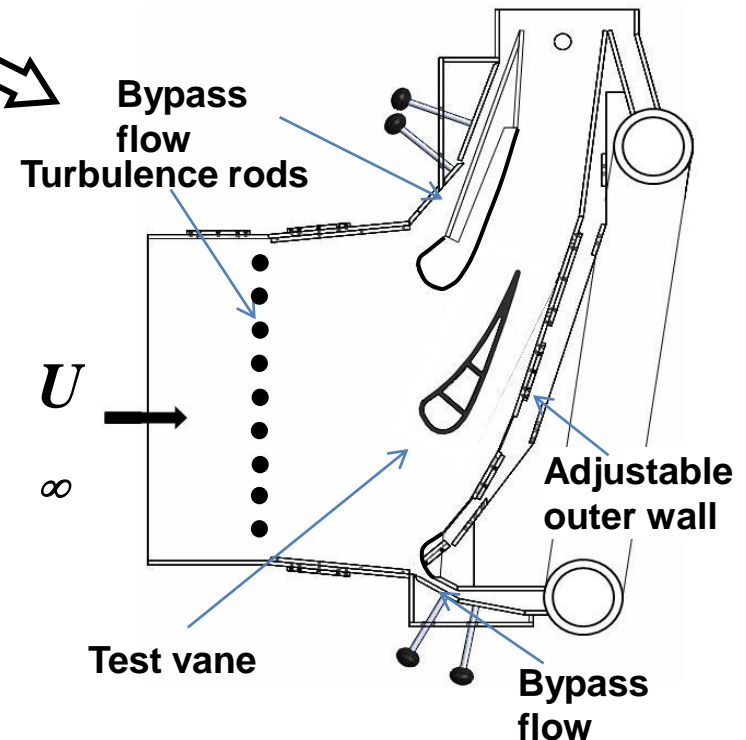
## Project Objectives, continued

- d. Develop the knowledge needed to design film cooling configurations on contoured endwalls.
- e. Perform detailed velocity and thermal field measurements along the vane, endwall, and downstream wake, with and without film cooling, to provide benchmarks to evaluate CFD simulations.
- f. Develop improved cooling designs specifically for the vane-endwall junction including mitigation of deposition effects.

# Simulated vane test facility at UT



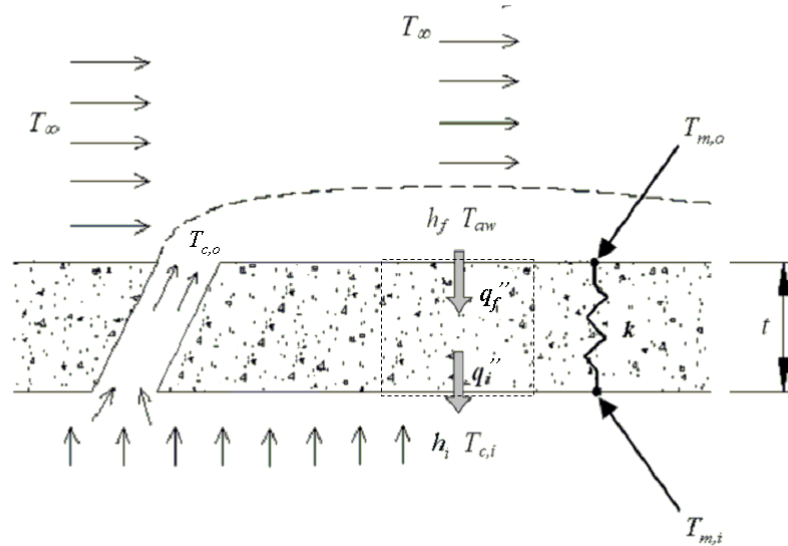
- Simulated Three Vane –Two Passage Linear Cascade
- Liquid nitrogen cooled secondary flow during overall effectiveness measurements
- Constant heat flux boundary condition on near adiabatic airfoil for heat transfer coefficient measurement
- Surface temperatures measured using FLIR P20 and P25 IR cameras



**Technical Approach: use matched Biot number models to obtain overall effectiveness which includes internal cooling effects**

**Heat transfer through the wall:**

$$q_0'' = \frac{T_{aw} - T_{c,i}}{\frac{1}{h_f} - \frac{t}{k} - \frac{1}{h_i}}$$



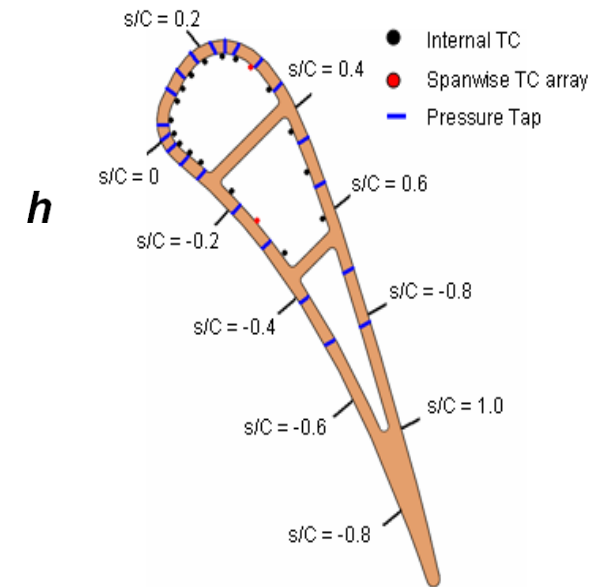
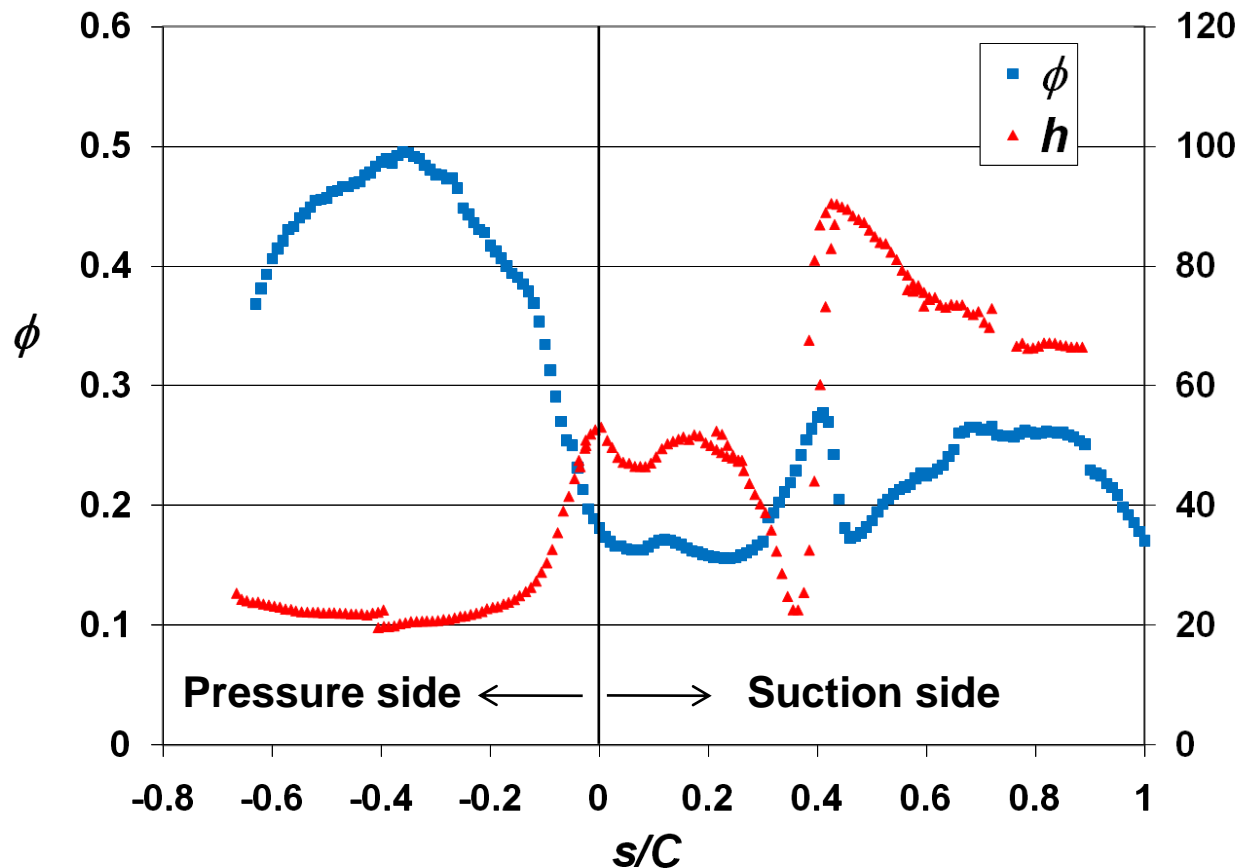
**Overall effectiveness:**

$$\phi = \frac{T_{\infty} - T_w}{T_{\infty} - T_{ci}} = \frac{1 - \eta}{1 + Bi + \frac{h_f}{h_i}} + \eta$$

**Experimental model needs to match  $Bi$  and  $h_f/h_i$**

A baseline was established by measuring  $\phi$  with internal cooling only.

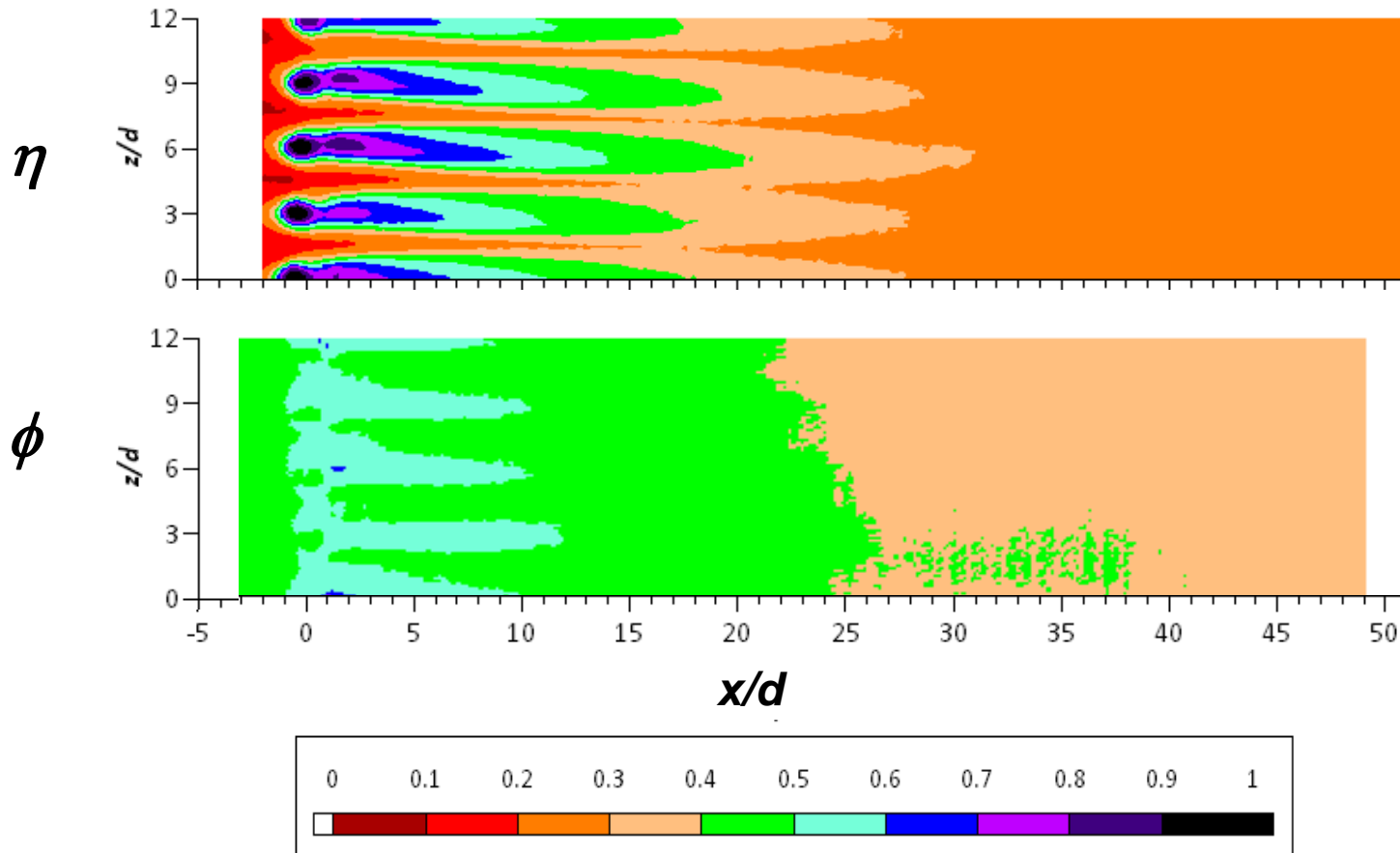
From Dees' dissertation (2010)



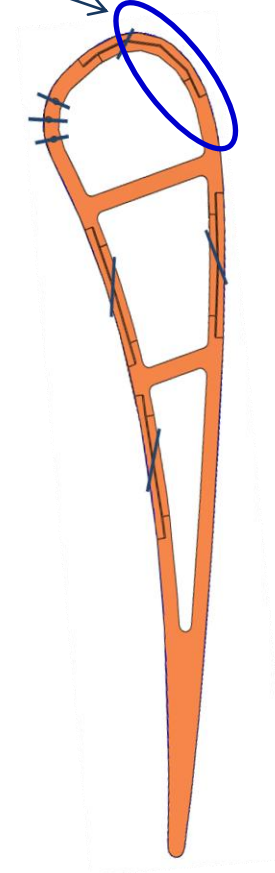
Variation in  $\phi$  was found to correlate with variation in external heat transfer coefficient.

# Comparison of $\eta$ and $\phi$ contours with film cooling shows redistribution of localized cooling by coolant jets

From Dees' dissertation (2010)



Region of measurements

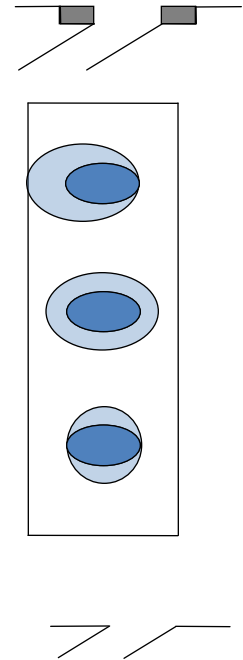
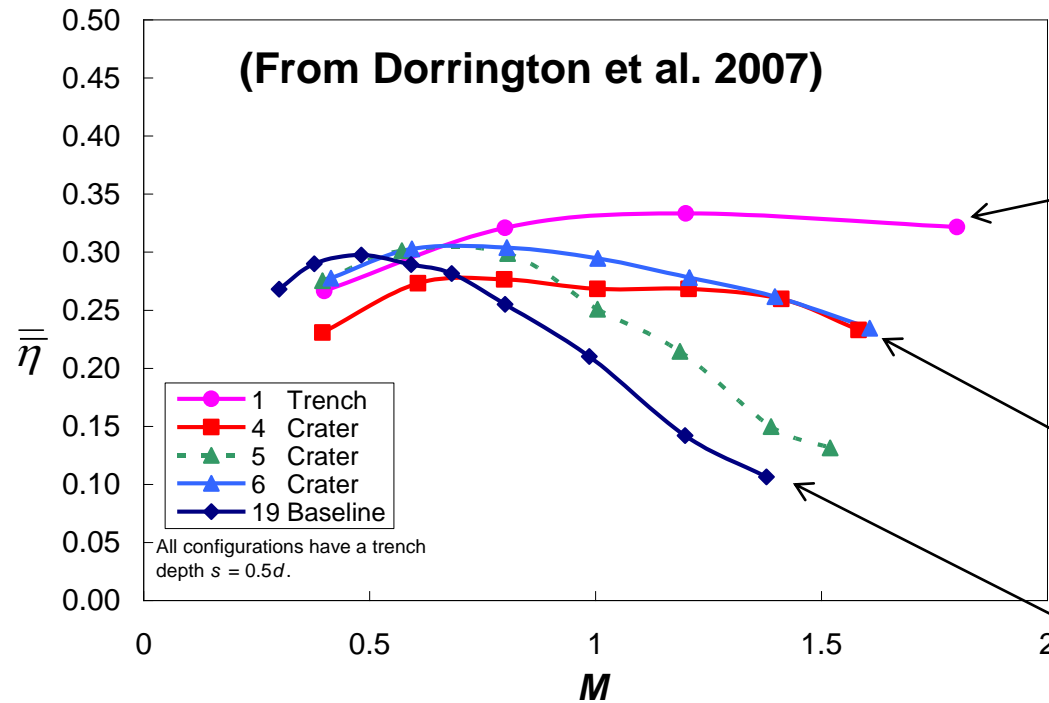


$$\eta = \frac{T_{aw} - T_{\infty}}{T_{c,e} - T_{\infty}}$$

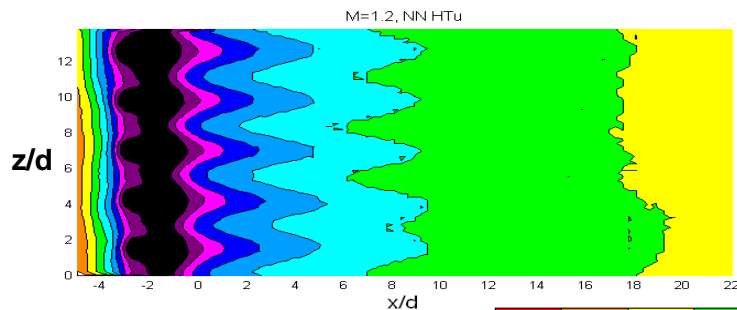
$$\phi = \frac{T_w - T_{\infty}}{T_{c,i} - T_{\infty}}$$

# This study will evaluate trench and crater film cooling configurations

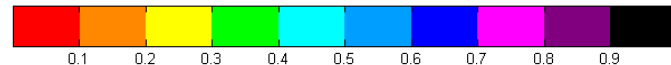
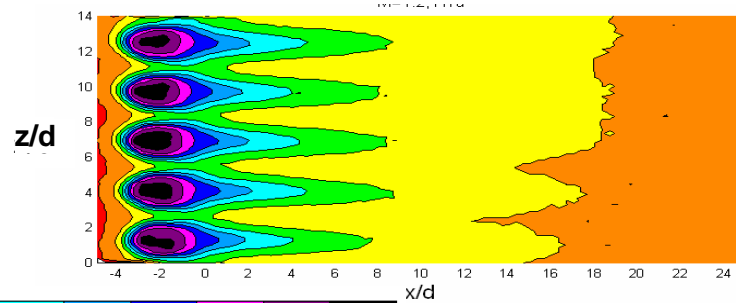
## Configurations



## Trench



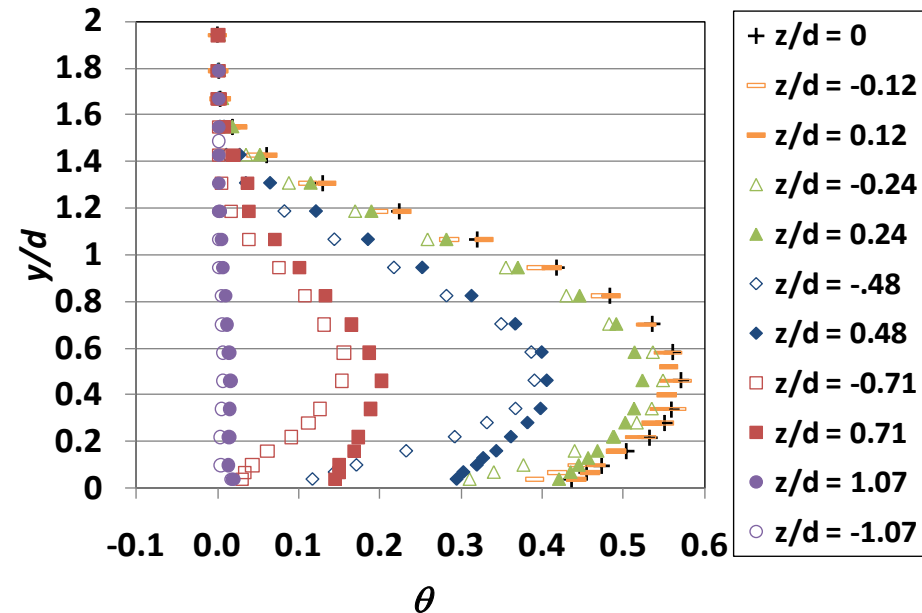
## Craters





# Thermal field measurements will be an important part of this study to show coolant jet details and conjugate heat transfer effects

## Adiabatic wall

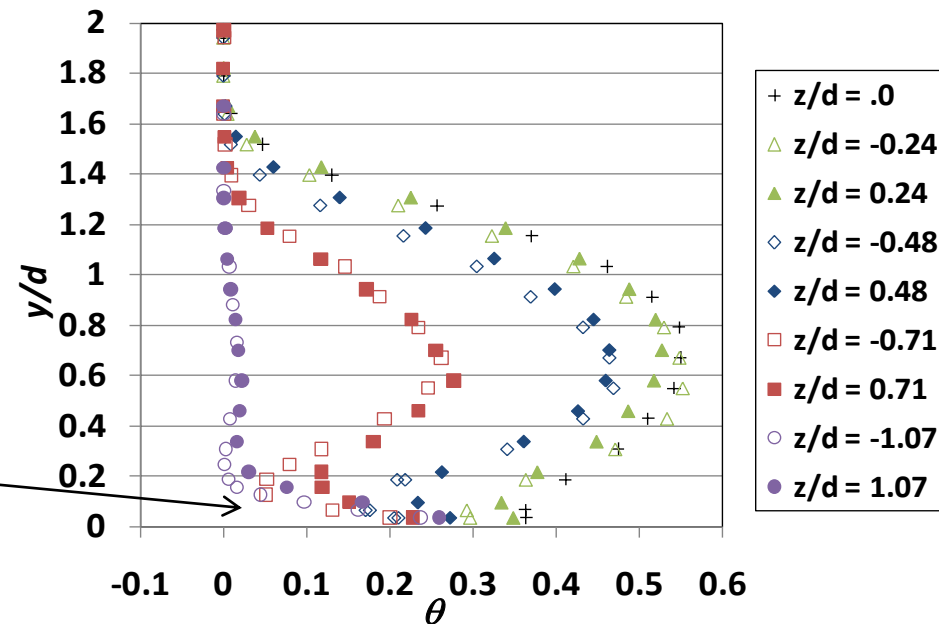


Note thermal boundary layer development

From Dees' dissertation (2010)

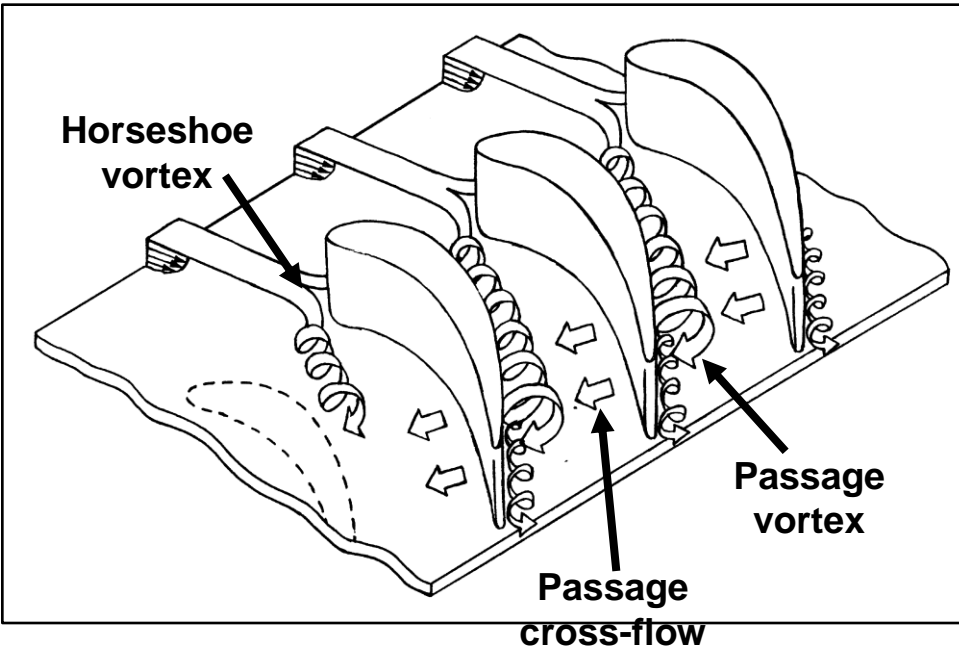
$$I = 0.75, x/d = 5$$

## Conducting wall

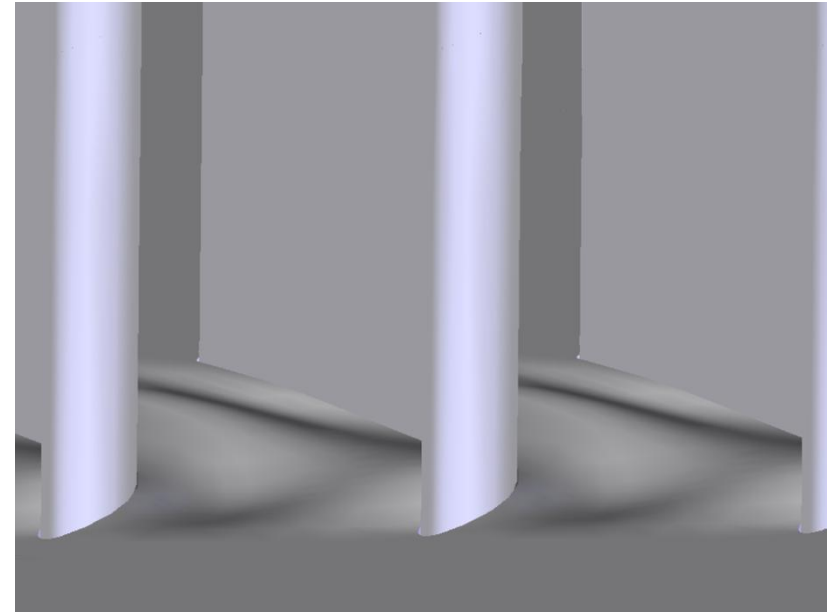


# The objective of our studies is to understand the effect of deposition on non-axisymmetric contoured endwalls

Langston [1980]

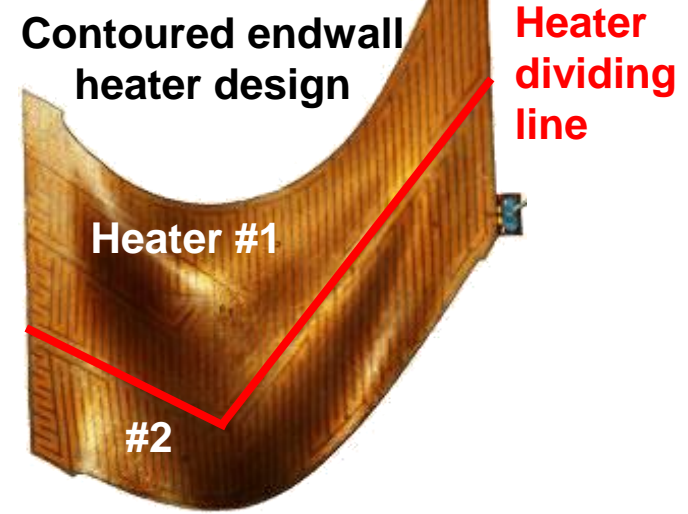
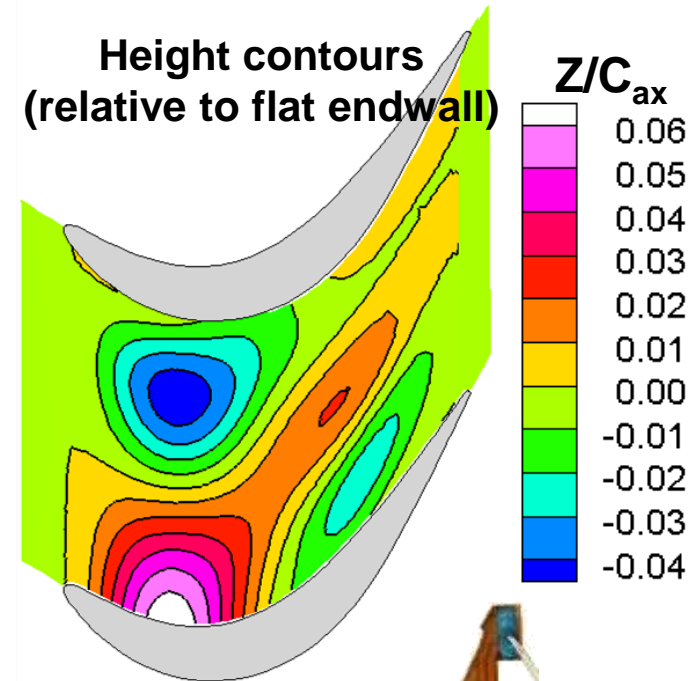
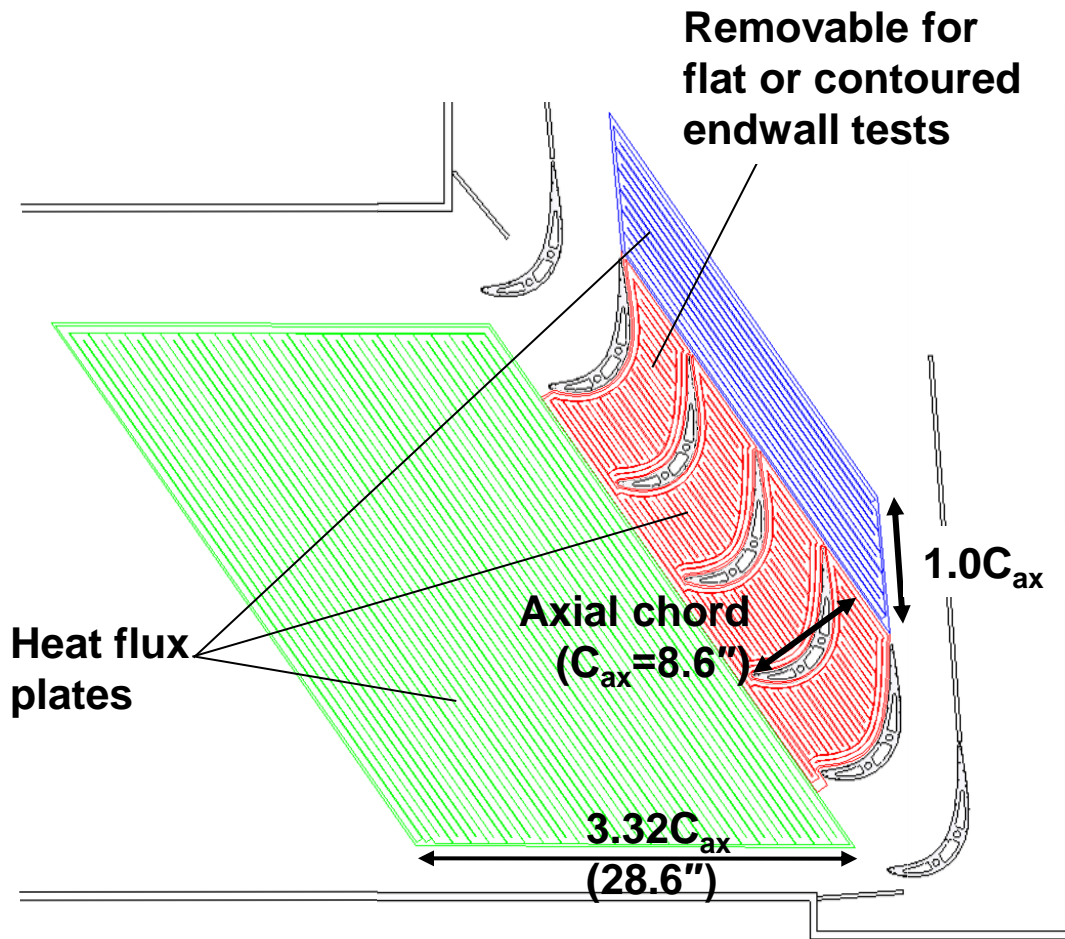


Contoured endwall

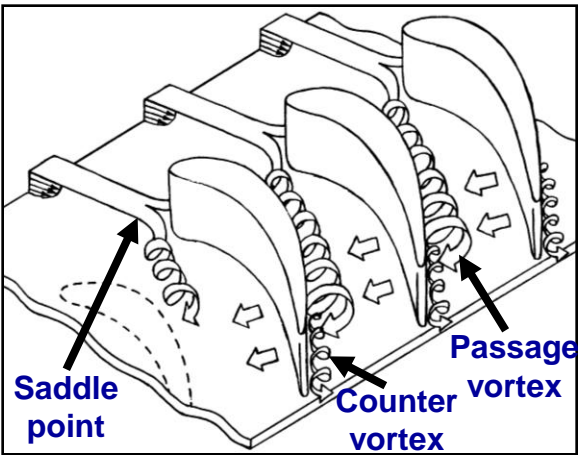


**Do deposits adversely affect heat transfer and film cooling with contoured endwalls?**

**A novel heater design was developed for contoured endwalls to allow it to conform to the surface**

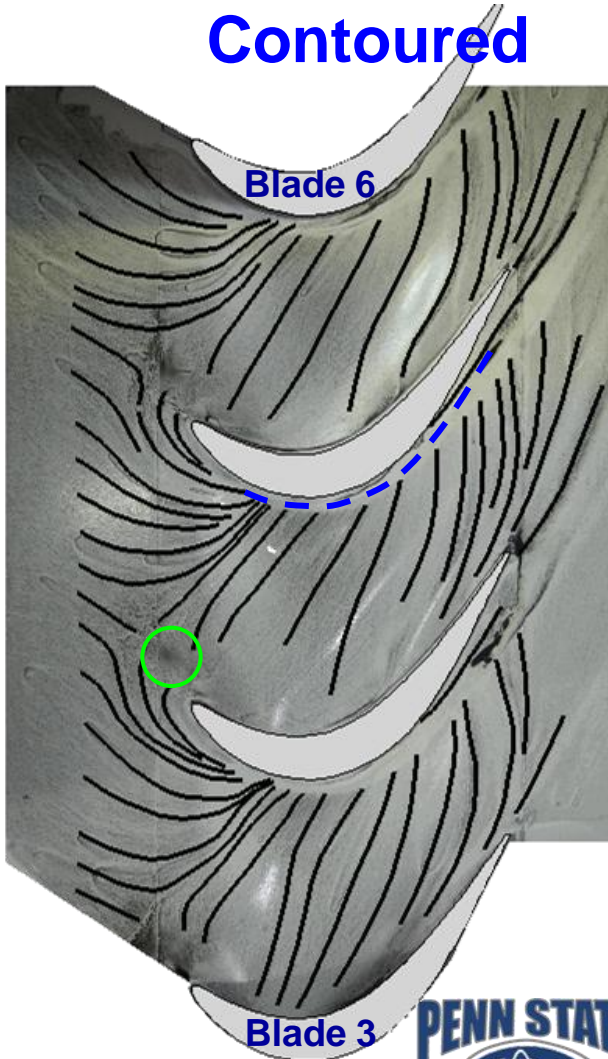
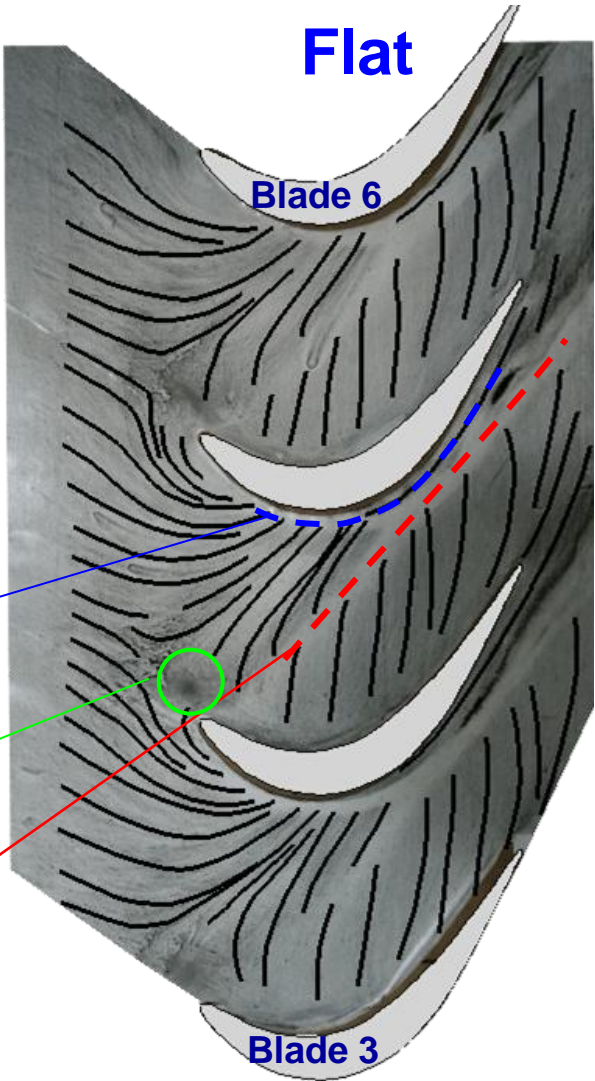


# Oil flow visualization for the contoured endwall did not show a scoured region caused by the passage vortex



Langston [1980]

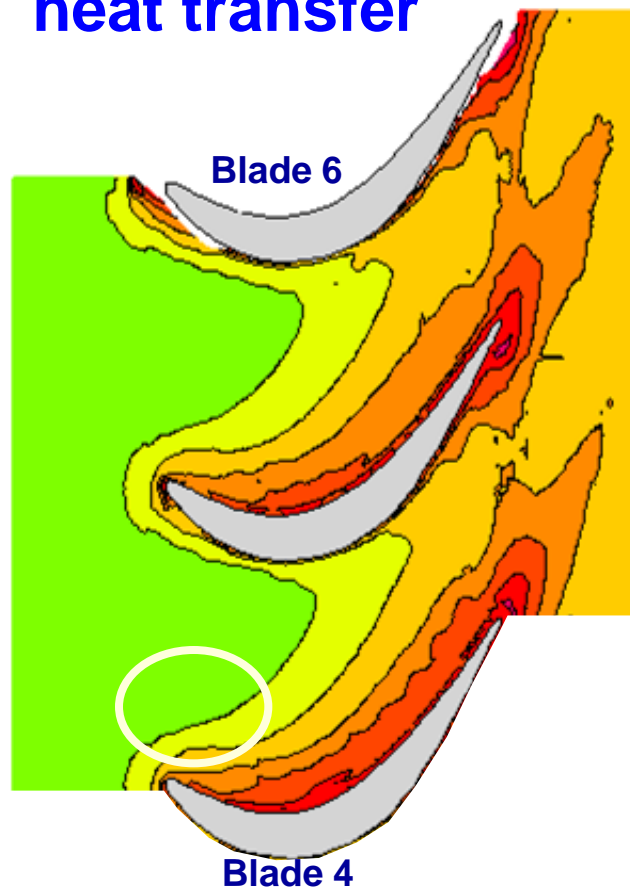
Counter  
vortex effect  
Saddle point  
region  
Passage  
vortex effect



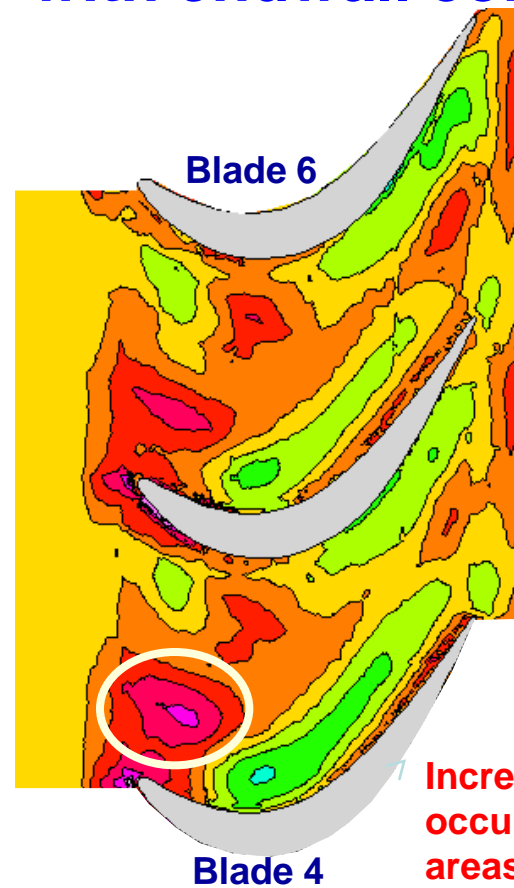


# Endwall contouring reduced heat transfer by 25% in the region of high heat transfer near the pressure side

## Flat endwall heat transfer



## Heat transfer augmentation with endwall contouring



Increased augmentation occurs in low heat transfer areas

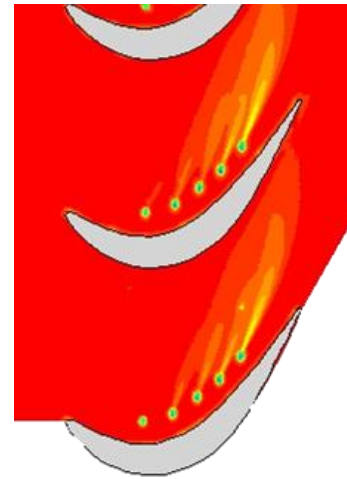
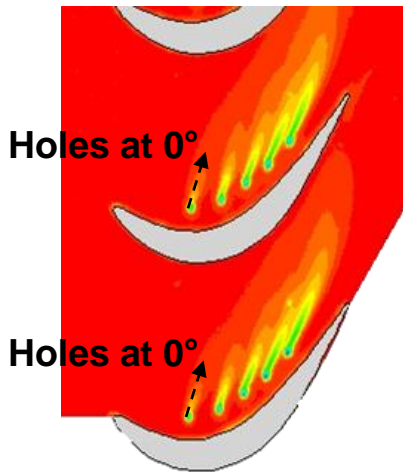
# Coolant does not sweep across the passage as well for the contoured endwall, compared to the flat endwall

Flat  
 $\overline{M}_{ideal}=1.0$

Contoured  
 $\overline{M}_{ideal}=1.0$

Flat  
 $\overline{M}_{ideal}=2.0$

Contoured  
 $\overline{M}_{ideal}=2.0$



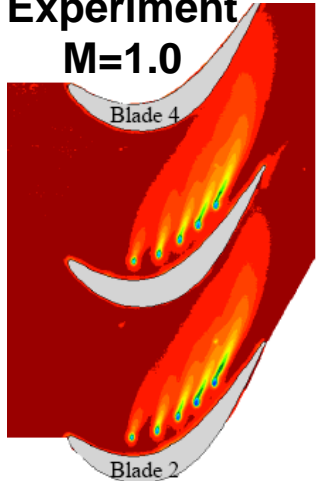
$$\overline{M}_{ideal} = \frac{\rho_j U_{j,ideal}}{\rho_\infty U_\infty} = \sqrt{\frac{\rho_j}{\rho_\infty} \left( \frac{P_{plen} - P_{s,endwall}}{P_{tot,in} - P_{s,endwall}} \right)}$$

$$\overline{M}_{ideal} = \sum_k (M_{ideal})_k / k \quad \begin{array}{l} j=\text{coolant} \\ k=\text{individual hole} \end{array}$$

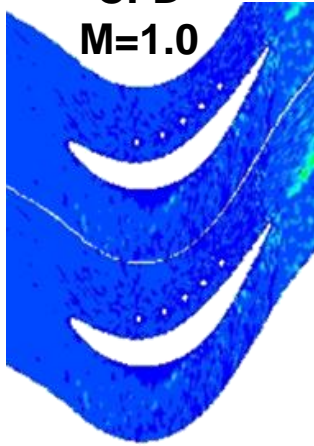
# PSU will investigate the effects of deposition on film-cooling with endwall contouring

## Flat Endwall

Baseline  
Experiment  
M=1.0

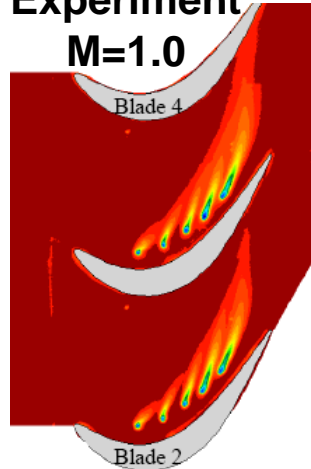


CFD  
M=1.0

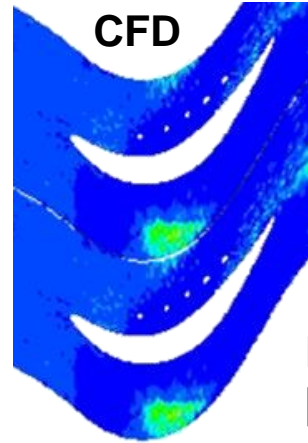


## Contoured Endwall

Baseline  
Experiment  
M=1.0

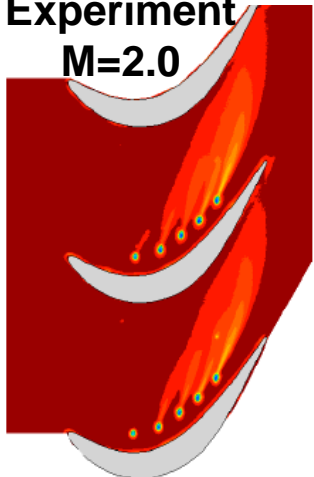


CFD

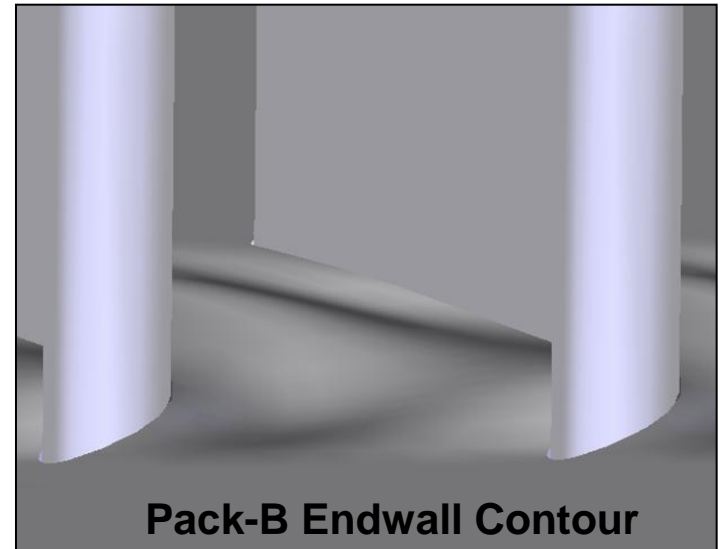


Experiments by  
[Lynch et al., 2010]

Baseline  
Experiment  
M=2.0



Baseline  
Experiment  
M=2.0



# Phases and Deliverables for the project

Phase	UT Deliverables	Penn State Deliverables
<b>Phase I Planning and Project Coordination</b>	<ul style="list-style-type: none"> <li>• Agreed upon plan for frequency and mode for communication</li> <li>• Agreed upon test plan for initial phases of work</li> </ul>	
<b>Phase II - Deposition with No Film-Cooling</b>	<ul style="list-style-type: none"> <li>• Identify where vane deposits occur</li> <li>• Vane heat transfer with deposits</li> <li>• Measurements of aero-thermal BL</li> </ul>	<ul style="list-style-type: none"> <li>• Conducting contoured endwall model</li> <li>• Identify where endwall deposits occur</li> <li>• Contoured endwall heat transfer with deposits</li> <li>• Measurements of aero-thermal BL</li> </ul>
<b>Phase III - Deposition with Trenched Film-Cooling on Vane and Contoured Endwall</b>	<ul style="list-style-type: none"> <li>• <math>\eta</math> and <math>\phi</math> with trenched holes on vane</li> <li>• Evaluation of new trench / TBC design for vane</li> </ul>	<ul style="list-style-type: none"> <li>• <math>\eta</math> and <math>\phi</math> for film-cooling on a contoured endwall</li> <li>• Evaluation of trenched film-cooling on contoured endwall</li> </ul>
<b>Phase IV - Optimized Hole/Trench on Vane and Contoured Endwall</b>	<ul style="list-style-type: none"> <li>• Effectiveness measurements for an optimized hole shape / trench for vane</li> <li>• Results from numerous hole shape / trench hole designs for the vane</li> </ul>	<ul style="list-style-type: none"> <li>• Effectiveness measurements for the optimal hole shape / trench for contoured endwall</li> <li>• Measurements of aero-thermal boundary layers</li> </ul>
<b>Phase V - Mating of Optimized Hole/Trench on Contoured Endwall</b>	<ul style="list-style-type: none"> <li>• Measurements of aero-thermal BL's for hole shape / trench on vane</li> <li>• Measurements in vane wake with optimal hole shape / trench</li> </ul>	<ul style="list-style-type: none"> <li>• Evaluation of mating between vane and contoured endwall with overall effectiveness measurements using optimized hole shape / trench</li> </ul>



# *Questions?*

